



Metis Coronagraph Ground Calibration

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On behalf of the Metis Team

VI Metis Workshop – Gottingen, D, 21 November 2018



Outline



- Solar Orbiter/Metis
 - General introduction
 - Instrument description
- Calibration facility and setup
- Instrument Calibrations
 - Vignetting function (Flat-Field)
 - VLDA Photon Transfer Curve
 - VL & UV Imaging & Radiometric Performance
 - VL Polarimeter Performance
 - Stray-light Suppression Performance and internal occulter centering
- Conclusions



In short



It works!



solar orbiter

Metis Coronagraph



- Polarized VL imaging @ 580 - 640 nm
- UV HI Ly α imaging @ 121.6 ± 10 nm
- FoV ($1.5^\circ \cdot 2.9^\circ$ annular, $1.6 - 3.0 R_{\odot}$ @ 0.28 AU)
- Spatial sampling element ≤ 4000 km (20") @ 0.28 AU
- Time resolution ≥ 1 sec
- Simultaneous VL and UV imaging

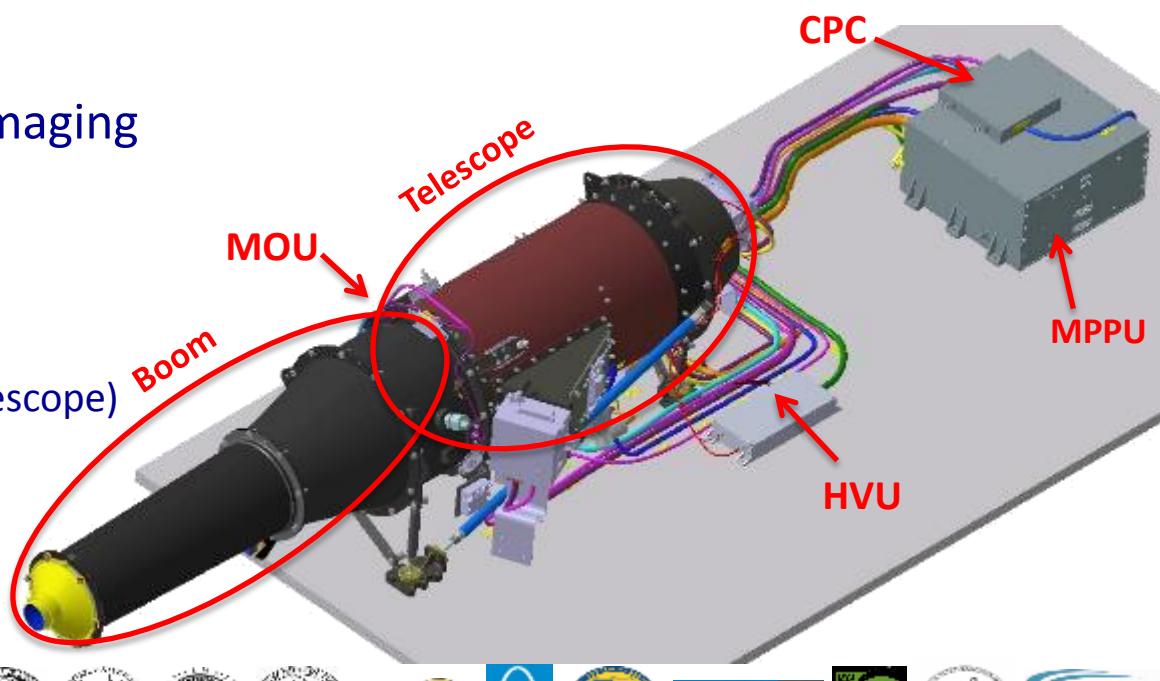
Units:

MOU METIS Optical Unit (boom and telescope)

MPPU METIS Power Processing Unit

HVU High Voltage Unit

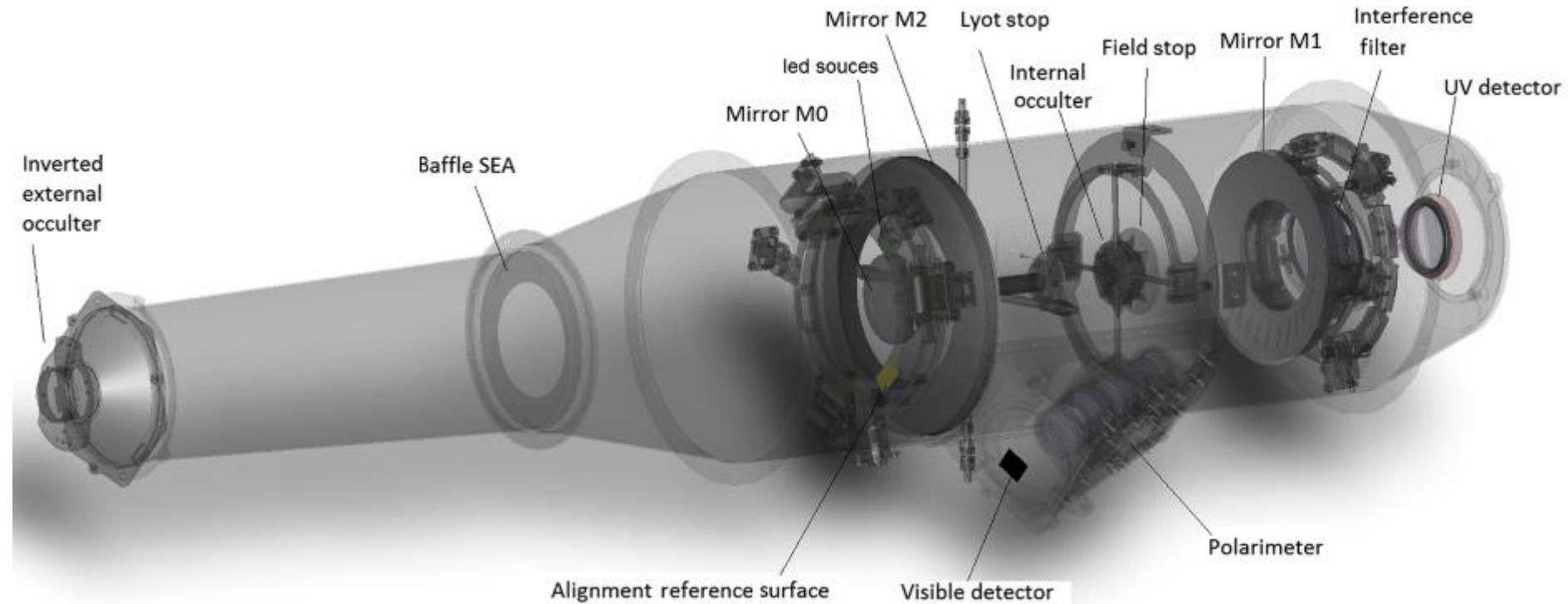
CPC Camera Power Converter





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Metis Opto-mech Layout



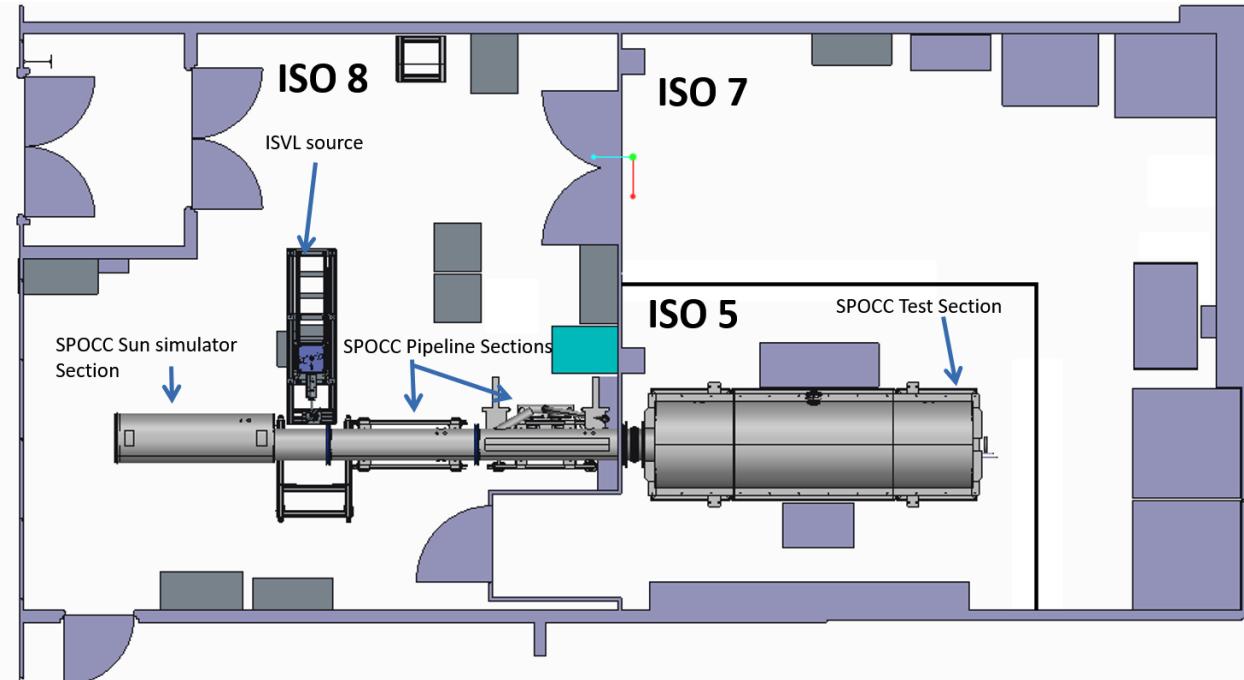


solar orbit

Ground Calibration Facility



- AIT/AIV and calibrations were performed at the OPSys (Optical Payload System) facility, an INAF laboratory hosted by ALTEC S.p.A. in Torino.
- OPSys: three communicating clean rooms which host the SPOCC (Space Optics Calibration Chamber)



- SPOCC is a solar divergence simulator that can operate in air and in vacuum.



Metis in the

Space Optics Calibration Chamber (SPOCC)





Metis in the

Space Optics Calibration Chamber (SPOCC)



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Ground Calibrations

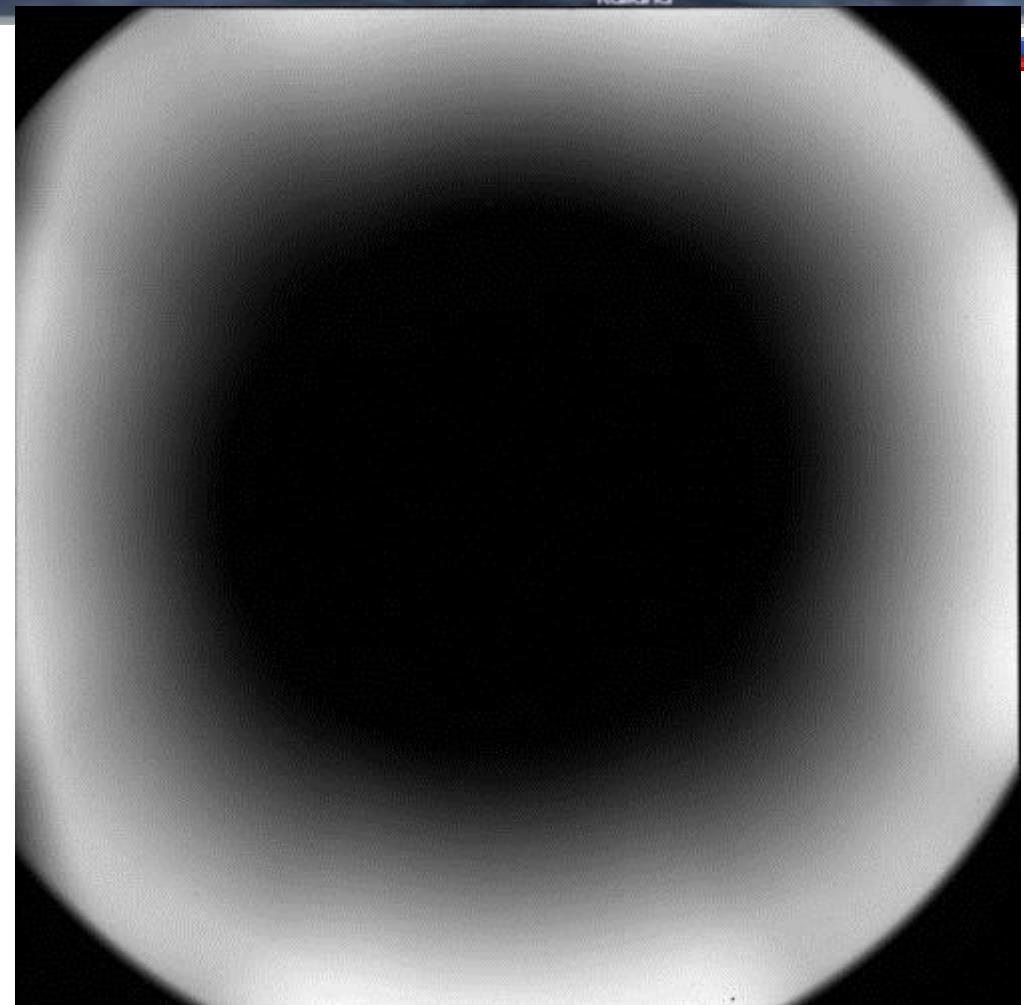


- Vignetting function (VL Flat-Field)
- VLDA Photon Transfer Curve
- VL & UV Imaging & Radiometric Performance
- VL Polarimeter Performance
- Stray-light Suppression Performance
- Internal occulter centering



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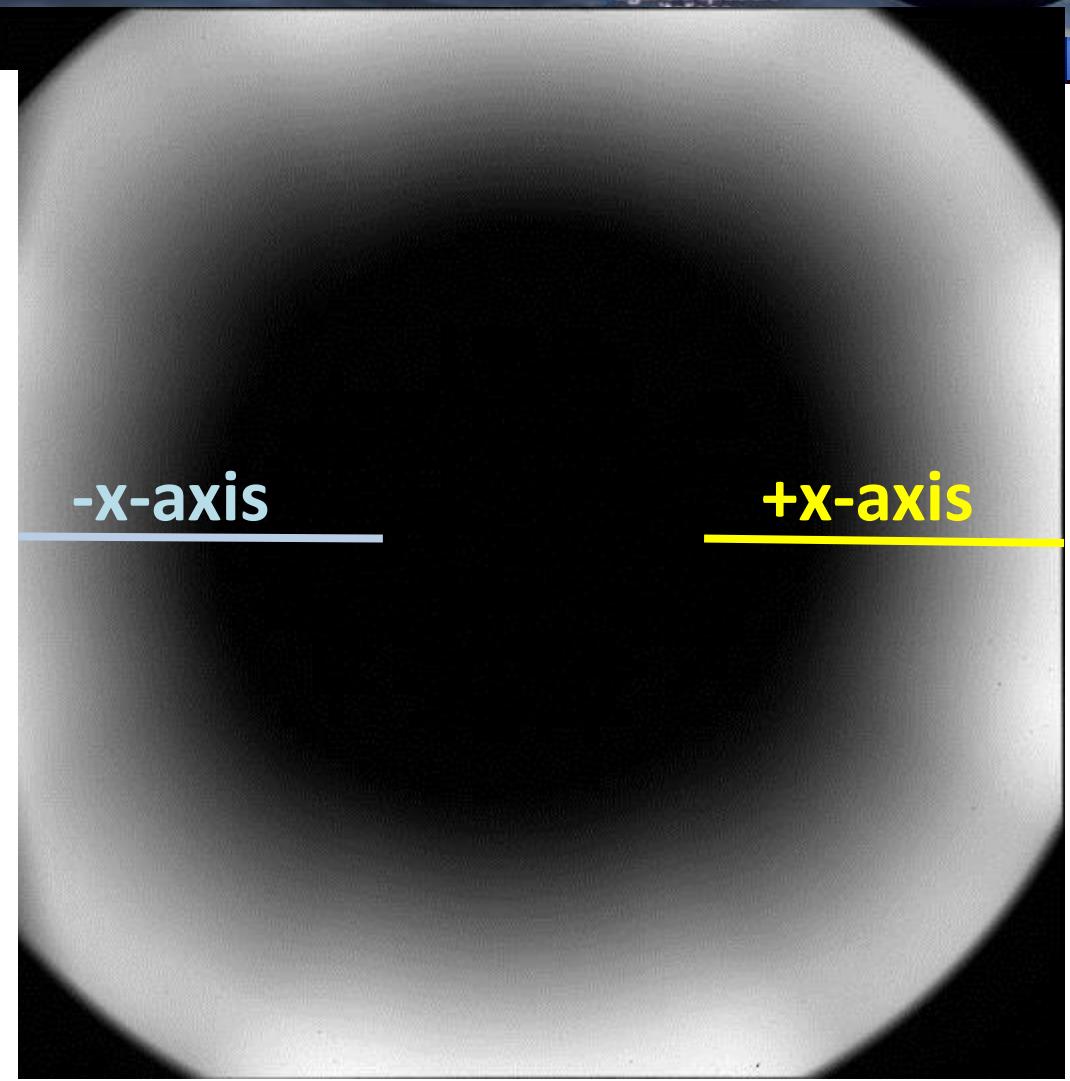
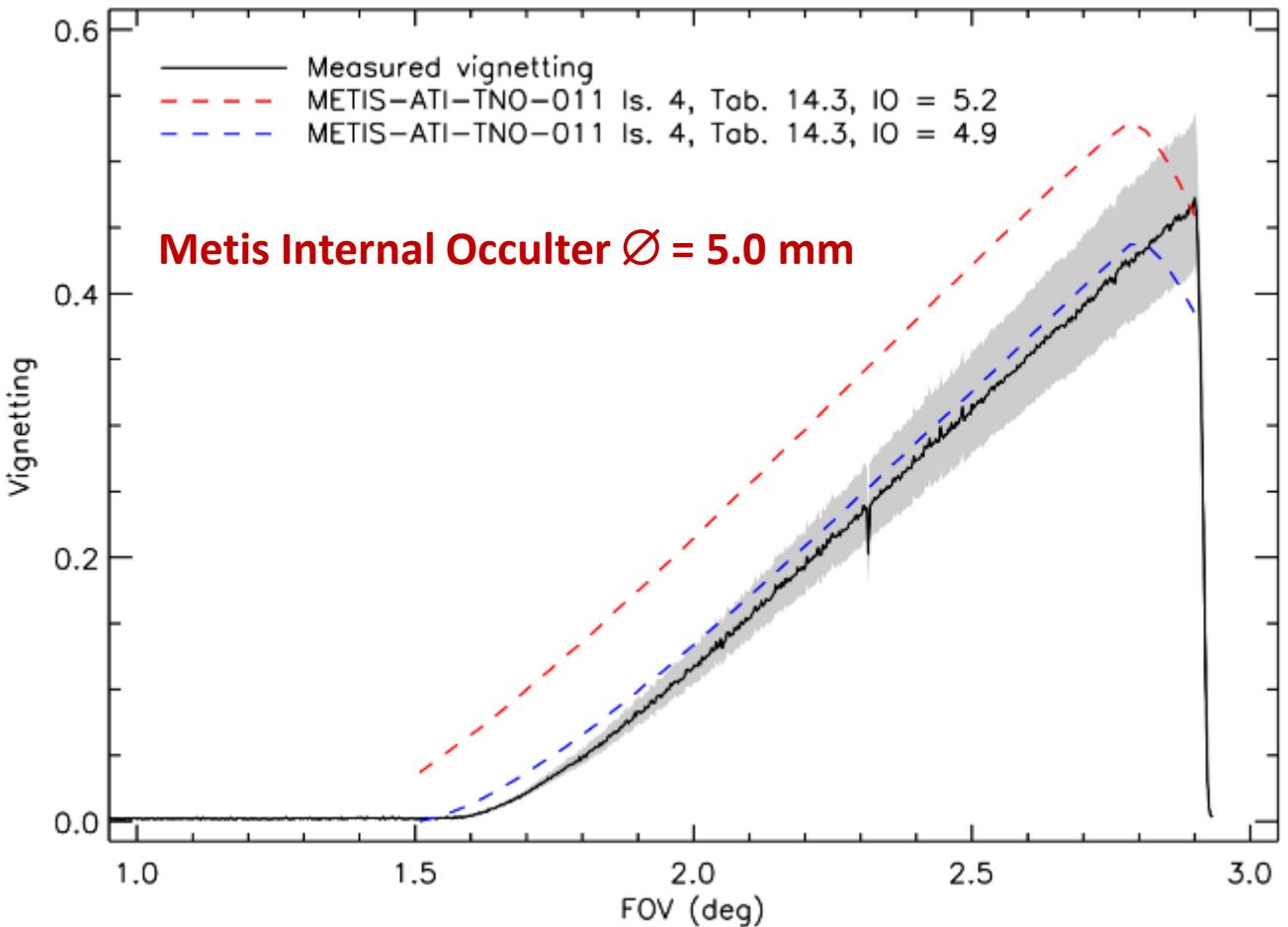
Vignetting Function & VL Flat Field





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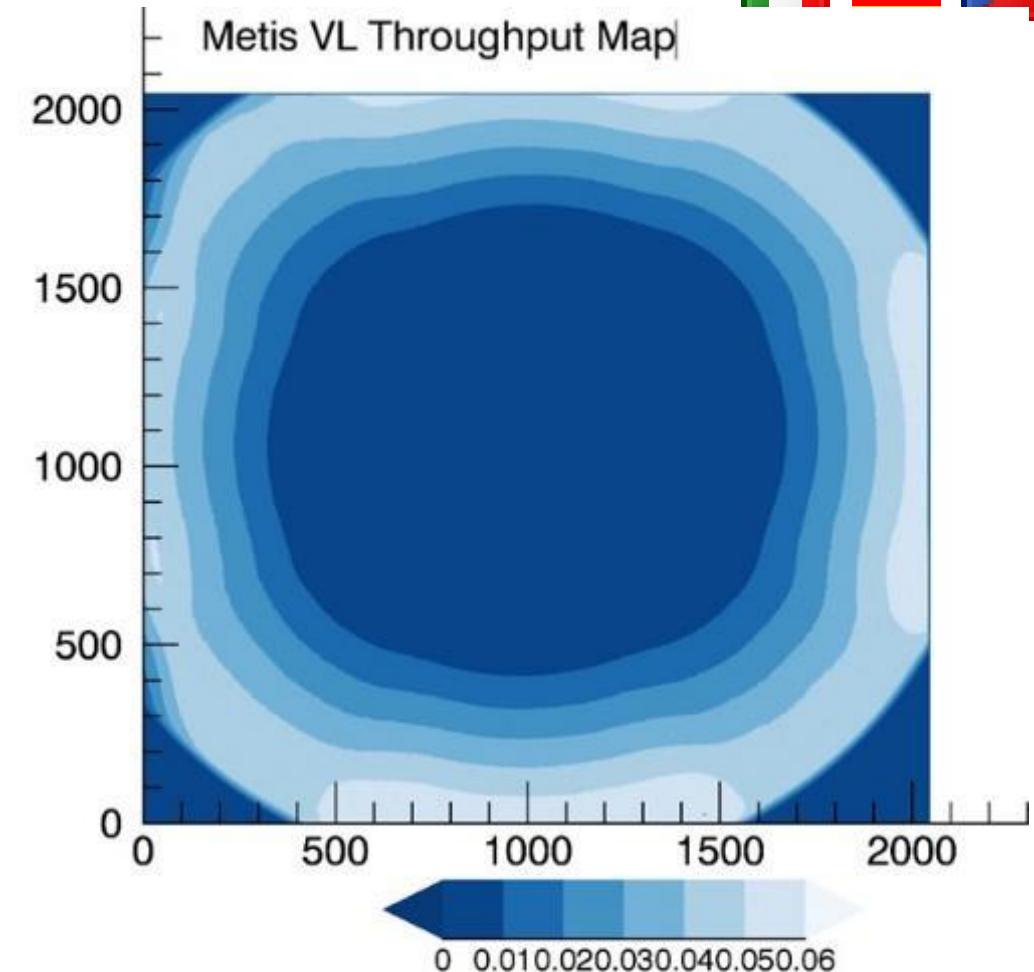
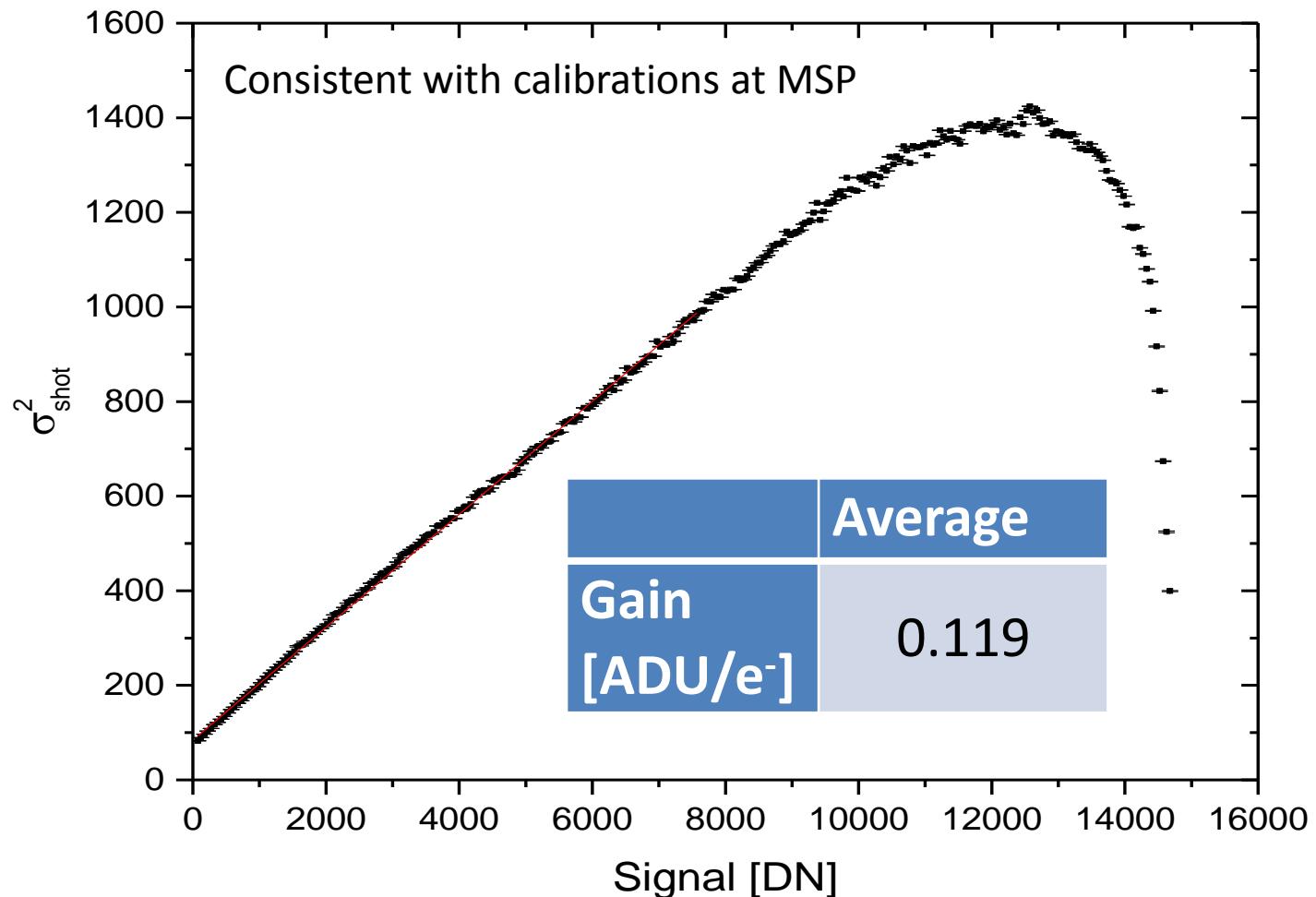
Vignetting Function & Flat Field





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VLDA Photon Transfer Curve





VL Measured Efficiency

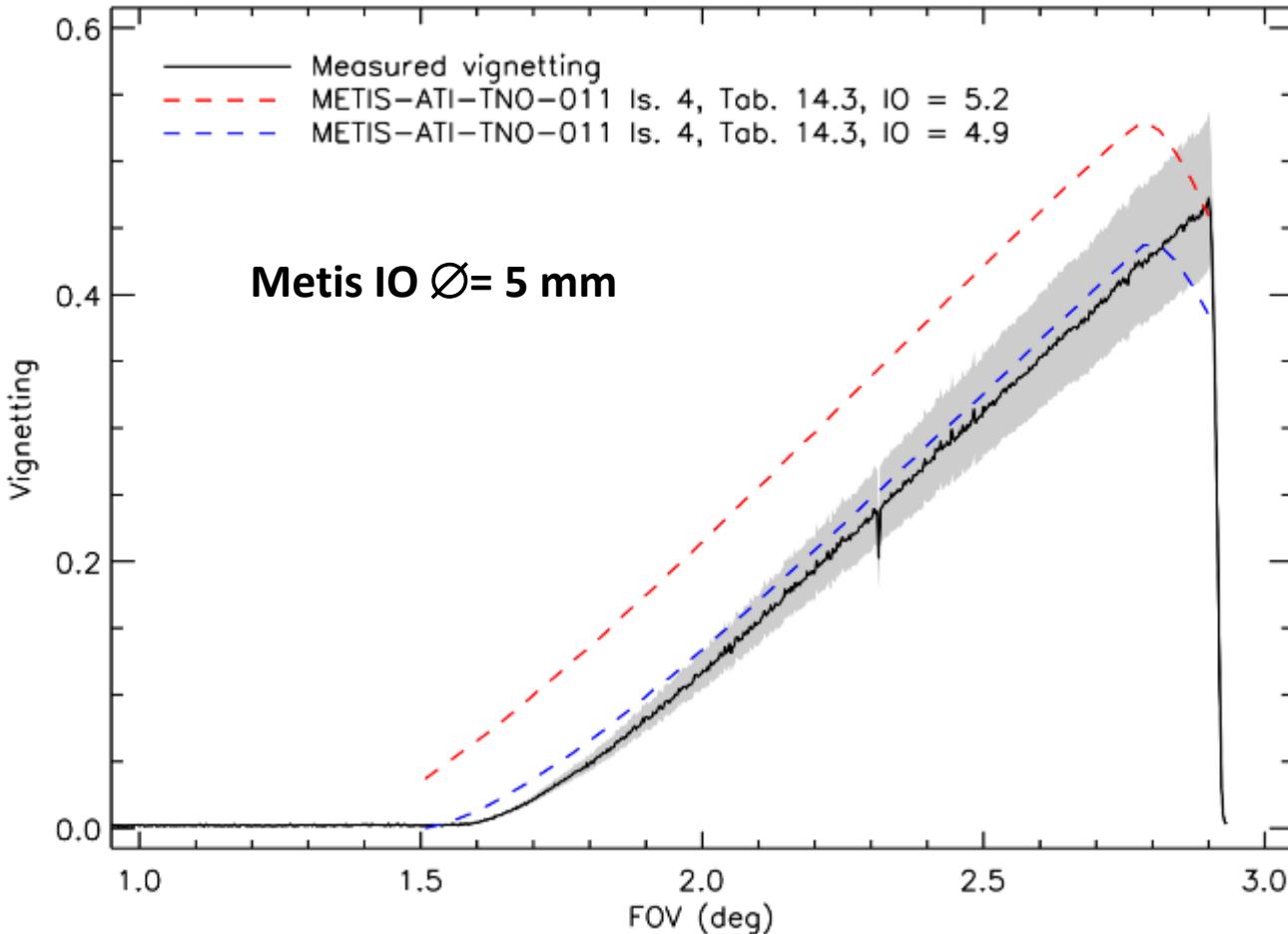


Visible-light			
Component	Efficiency	Error	Reference
M1	0.9	0.02	METIS-ANT-NCR-018 Appendix 1
M2		0.02	INF 28/07/2016
IFA	0.886	0.001	INAF Fineschi et al. 22/092016
Polarimeter	0.33	0.05	METIS- ATI-TNO-011 (estim.: 0.27)
			0.33 Measured on the EBB with Muller
QE	0.5	0.05	SpectroPolarimeter
Total efficiency	0.118	0.012	METIS-MPS-AT-03b-PFM-RP0005
Estim. Tot. Eff.	0.096		METIS- ATI-TNO-011 (Tab.14-1)



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VL Vignetting Normalization



Vignetting =
= Meas. Throughput [phot. detect./phot. in] / Efficiency
(Efficiency = 0.10 ± 0.01 from components measurements)

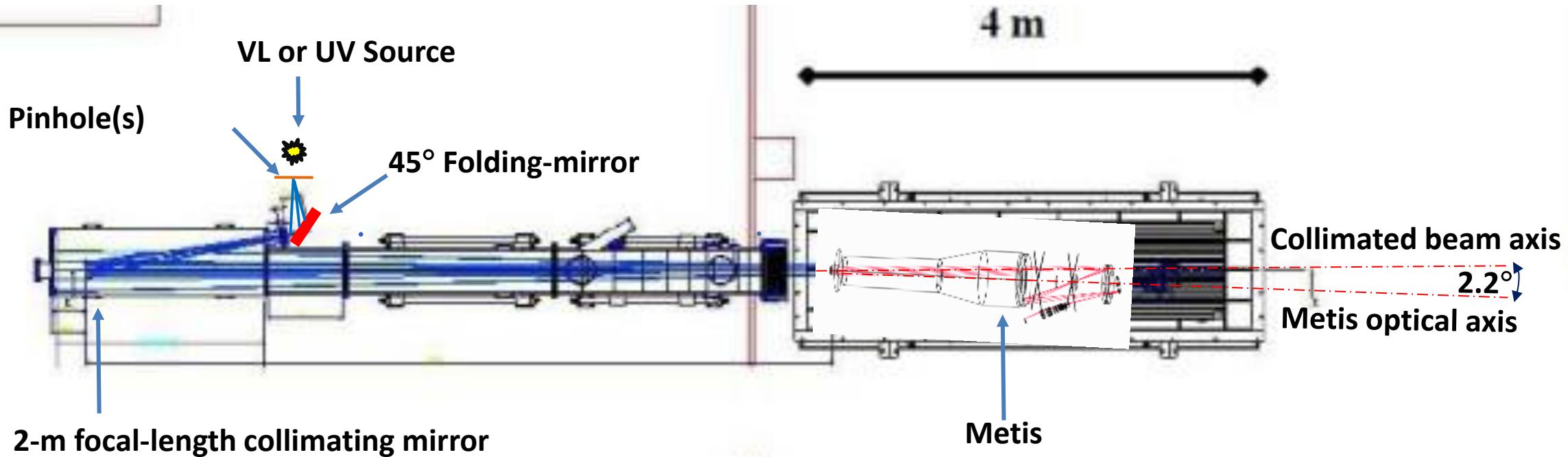


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Off-axis Metis in SPOCC



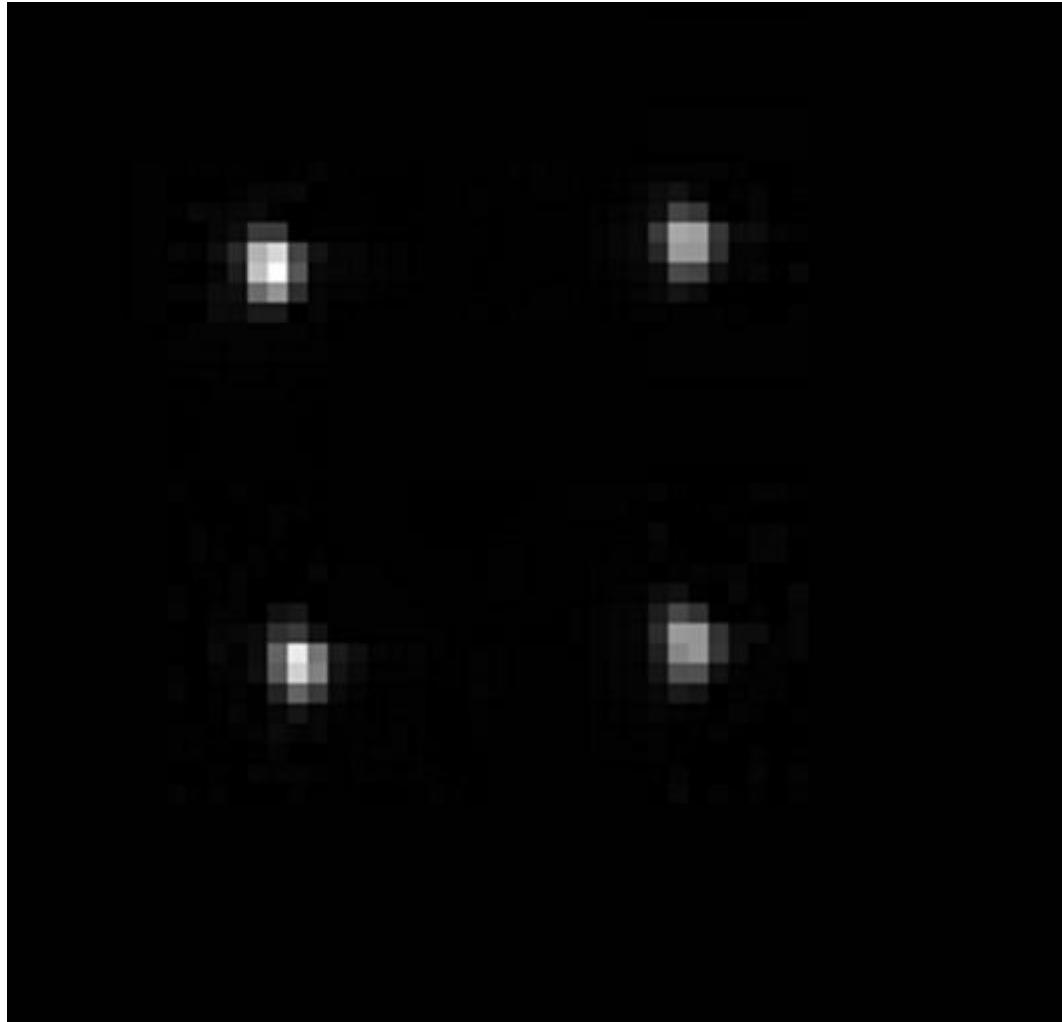
Metis set-up for VL and UV Measurements





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VL Imaging Performance



Target:

200- μm pinhole \times 4

SPOCC collimator f.l. = 2030 mm

VL effective f.l. = 200 mm

Demagnification = 0.098

simulating 4 “artificial stars”
with 20- μm FWHM on VLDA

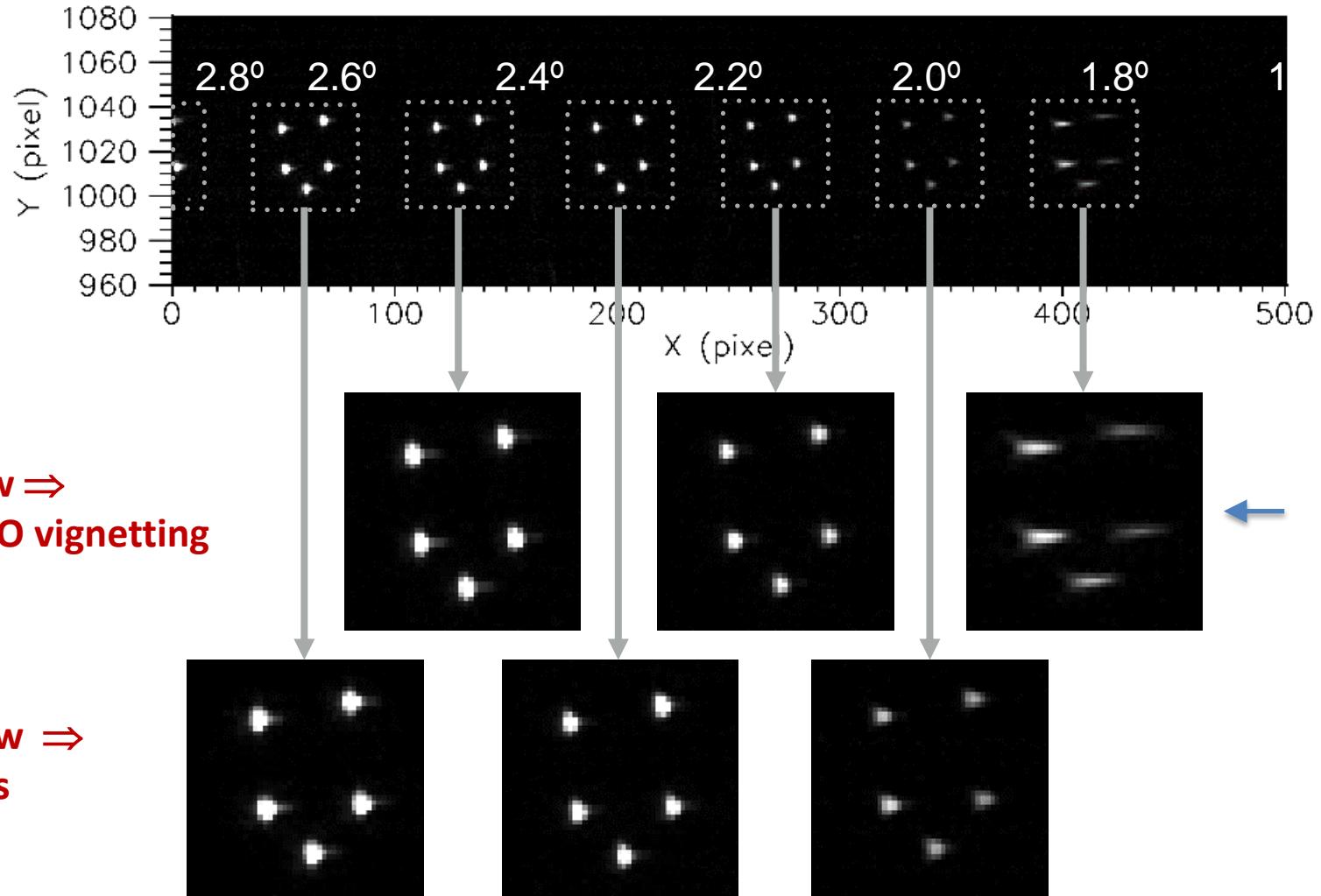


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VL Imaging Performance vs Field-of-View



- Inner Field-of-View ⇒ Diffraction from IEO vignetting
- Outer Field-of-View ⇒ Optical aberrations

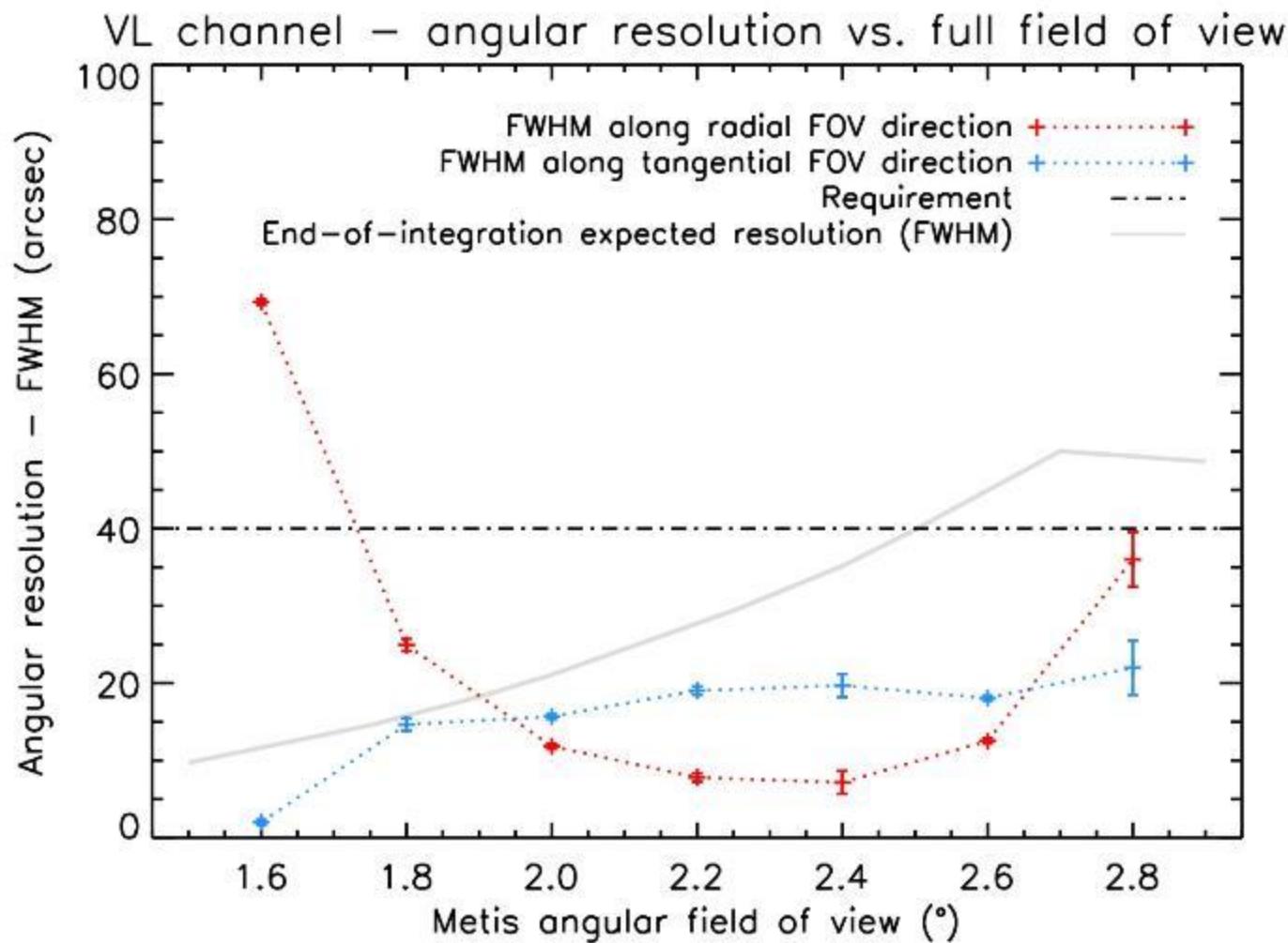


Natural diffraction
due to the highly
vignetted inner FOV



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VL Imaging Performance vs Field-of-View



VL Imaging Performance
↓
well within requirements



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VL Radiometric Response



Metis off-pointed at 2.2° sees an
“artificial Sun” at 1 AU

VL Radiometric Response =

VLDA signal (DN/s/pxl)/Input Radiance (W/cm²/sr)

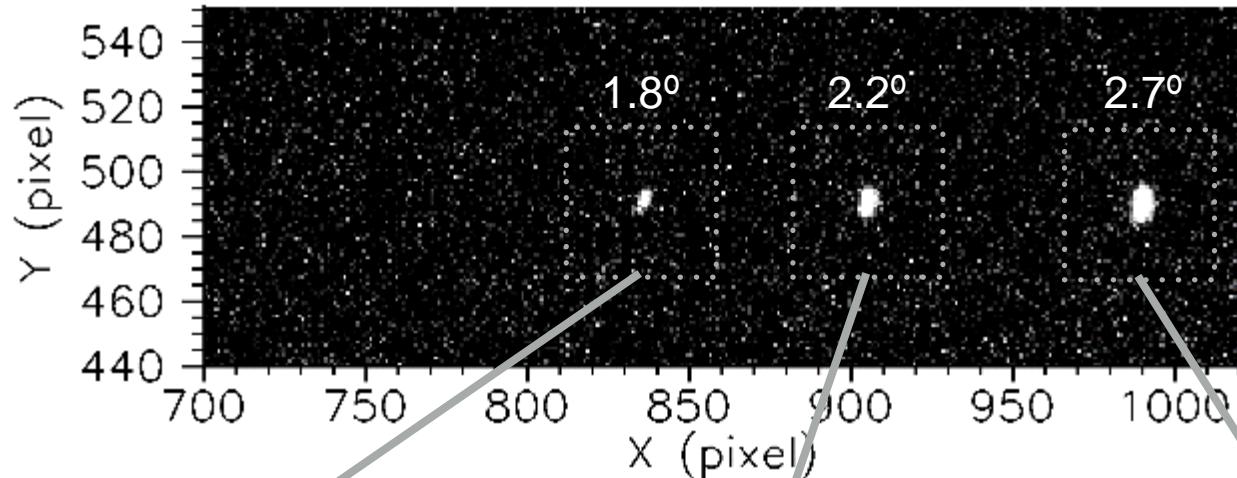
$$\begin{aligned} \text{VL Throughput @ FOV } 2.2^\circ &= \\ &= (1.9 \pm 0.2) \text{ e-2} \end{aligned}$$

(estimate based on component-level: 2.3 e-2)

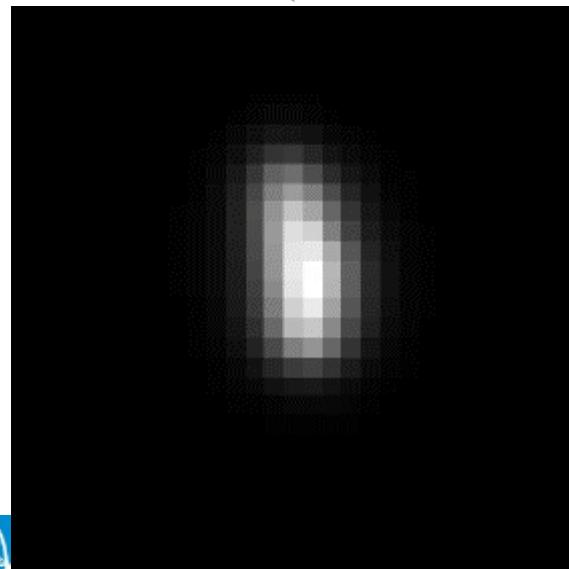
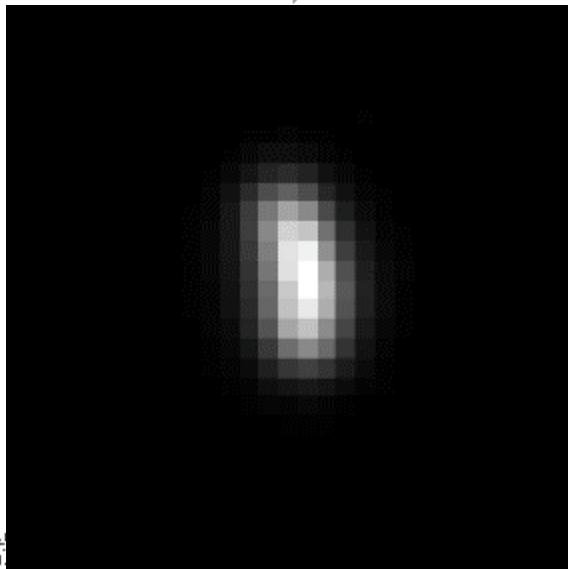
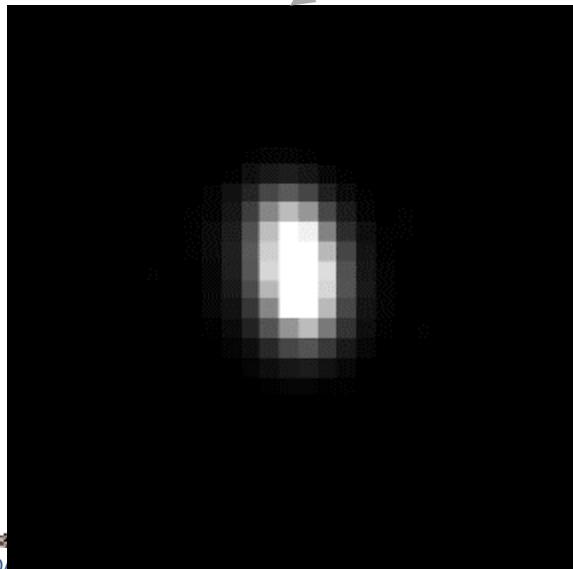




UV-analog Imaging Performance vs FoV



1.8° 2.2° 2.7°



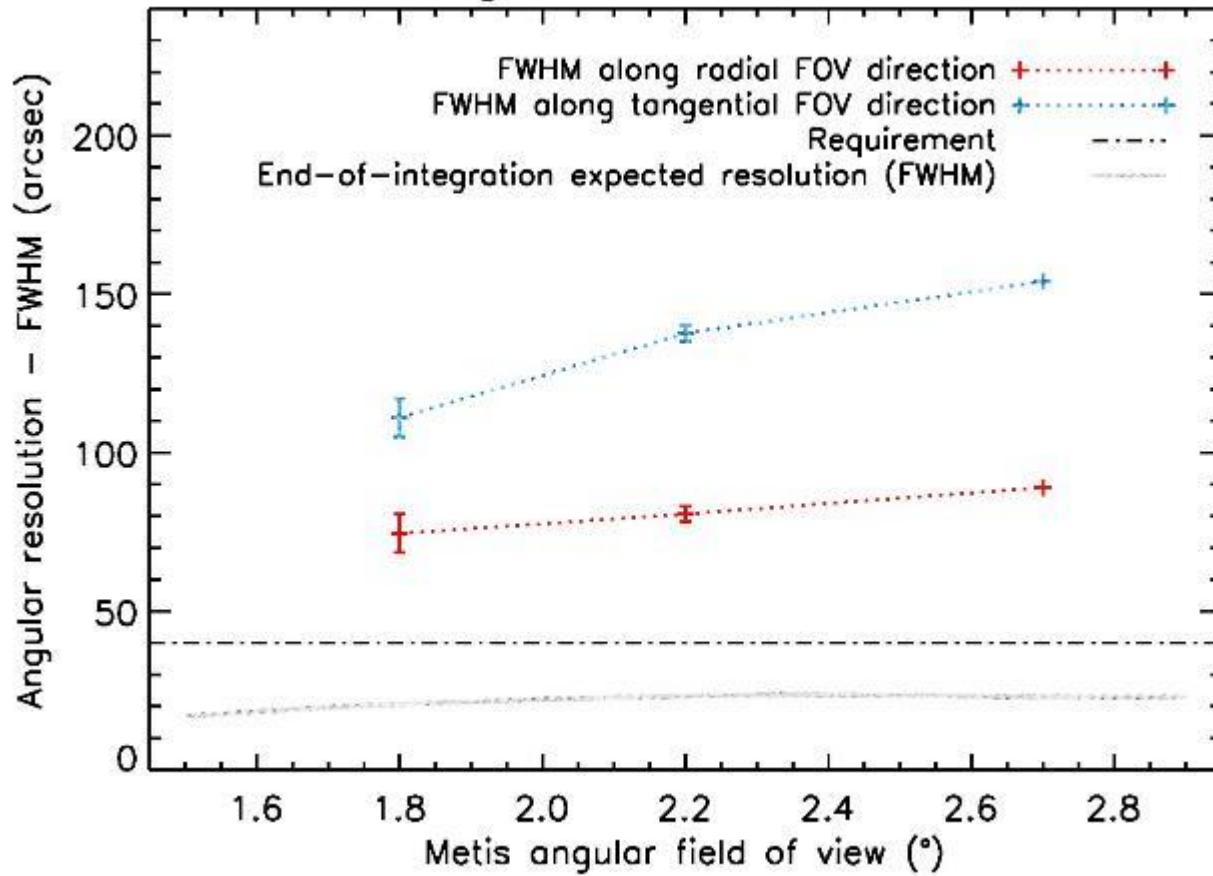


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UV-analog Imaging Performance vs FoV



UV channel – angular resolution vs. full field of view



The UV imaging performances, when the UVDA is operated in analog mode, are limited in spatial resolution to: 3-4 pixel \Rightarrow 80-100 arcsec

Cfr.

P62 F. Frassetto: Optical performance
P71 V. Da Deppo: Alignment procedure



UV Throughput Estimated from Components Measurements



UV (121.6 nm)

Component	Efficiency	Error	Reference
M1	0.54	0.12	Bear 02/10/2016 (T=0.77) &
M2	0.54	0.12	IFN 29/09/2016 (T=0.77); IFN 09/09/2017 (T=0.59); IFN 10/01/2017 (T=0.54)
IFA	0.24	0.04	INAF Fineschi et al. note 22/09/2016 Selected FM=0.24 (email by E. Antonucci Tue, 27 Sep 2016)
MgF ₂ window	0.18		MPS private communication
QE			
Total efficiency	0.013	0.004	
Efficiency w/o QE	0.07	0.02	0.36
Estim. Eff. w/o QE	0.16		
Eff. Estim./Meas.	2.27		



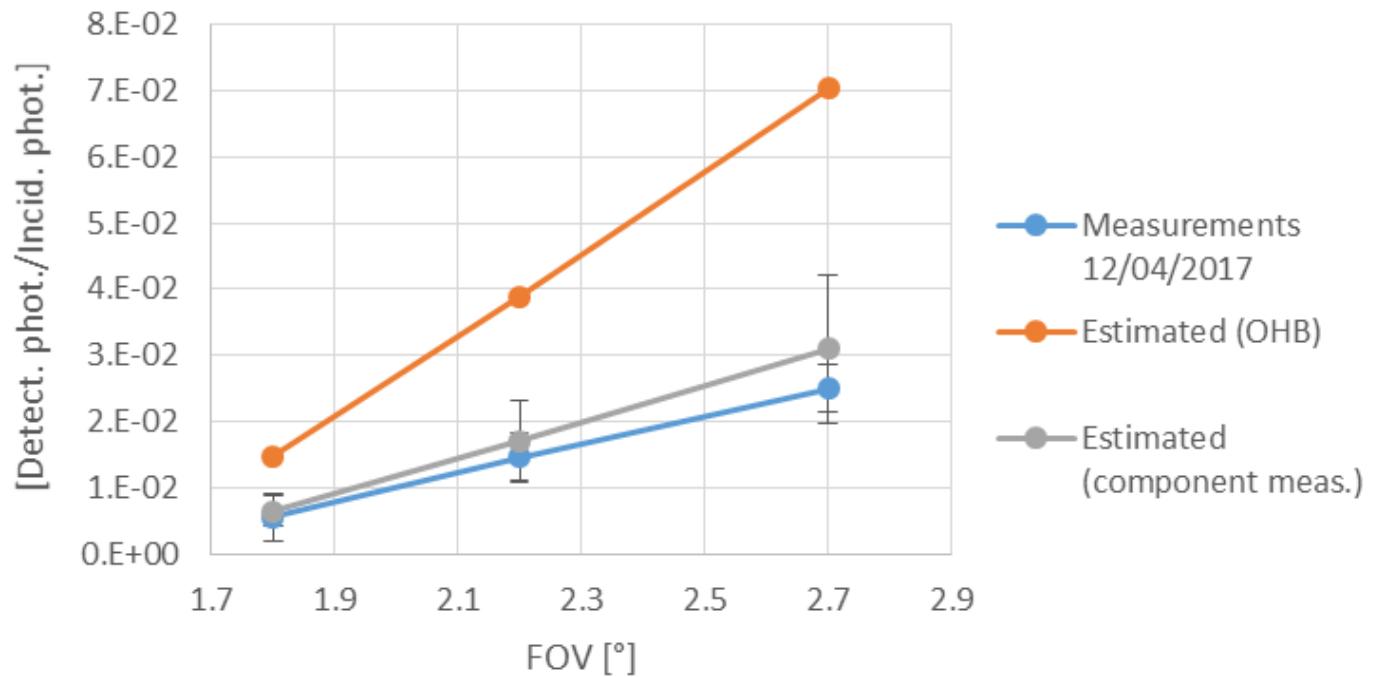
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UV Throughput

E2E Measurements vs Estimated from Components Measurements



UV Throughput (efficiency & vignetting) w/o Gain

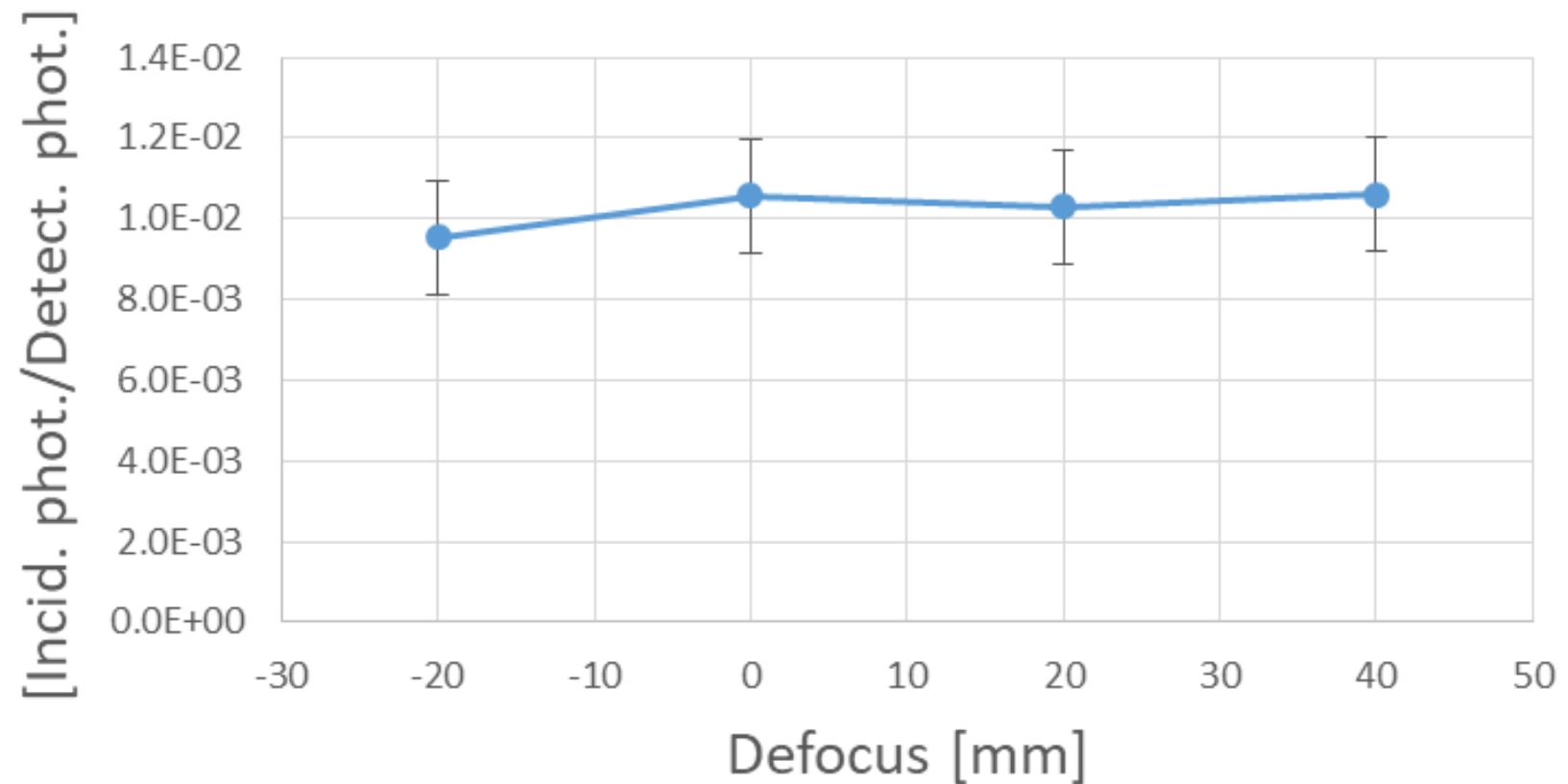


$$\text{Response Efficiency [DN/phot.In]} = (6.6 \pm 0.1)\text{e-1 [DN/s/pxl/phot.In]}$$



UV Throughput Measured during Defocus Test

UV Throughput at 2.2° w/o Gain





UV Response Efficiency



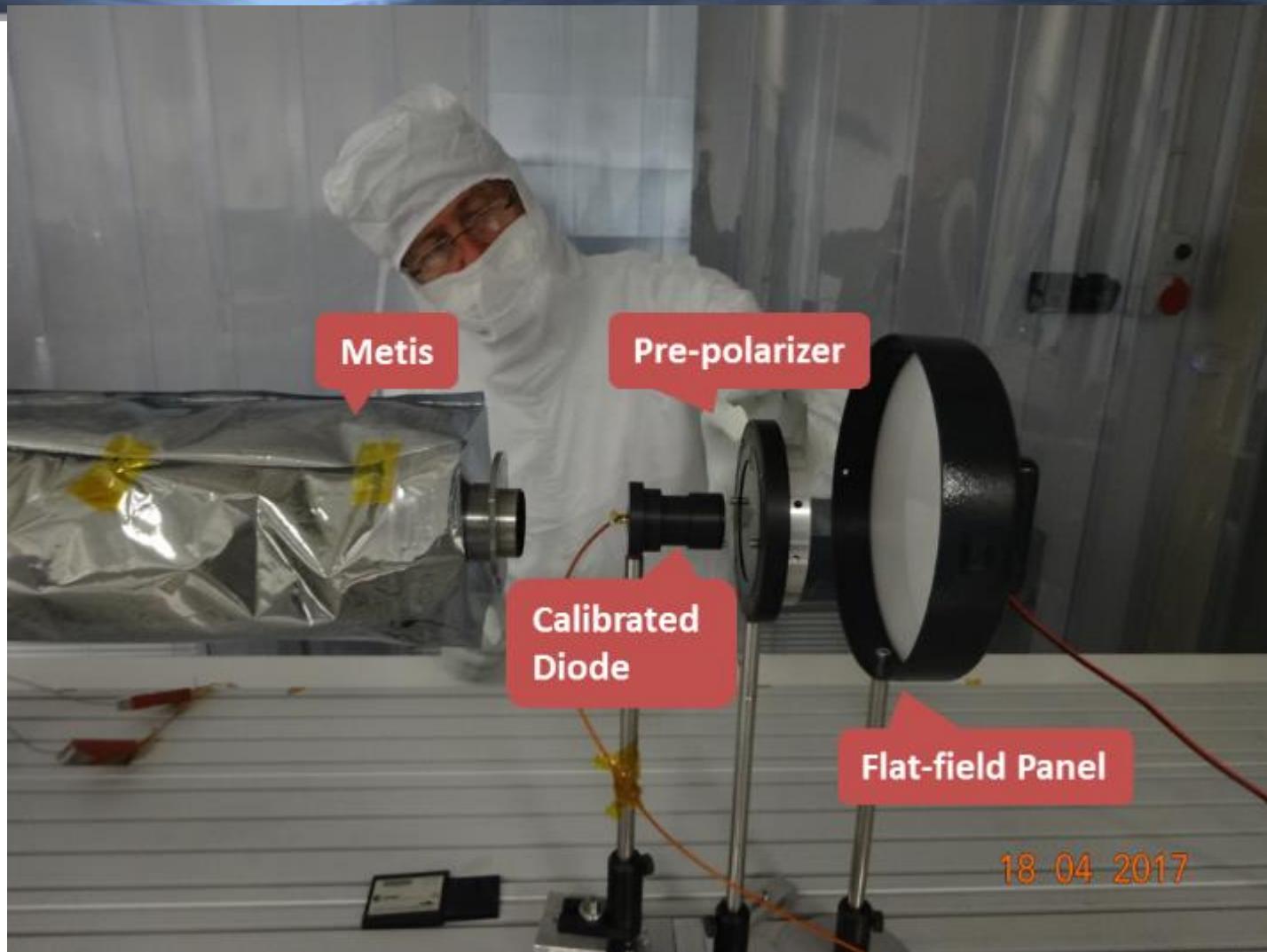
Response Efficiency = **(6.6 ± 0.1)e-1 [DN/s/pxl/phot.In] =**

$$= \frac{\text{Response Throughput [DN/s/pxl/phot.In] (measured @ FoV} = 1.8^\circ, 2.2^\circ, 2.7^\circ}{\text{Vignetting@ FoV} = 1.8^\circ, 2.2^\circ, 2.7^\circ \text{ (derived from VL radiometric calibration)}}$$



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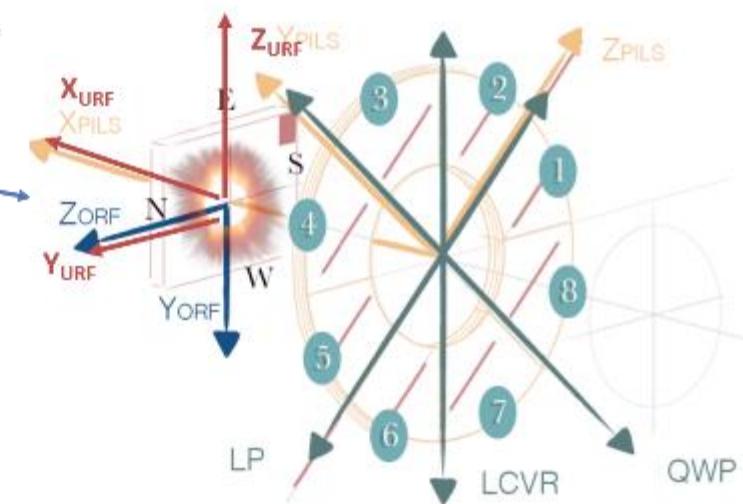
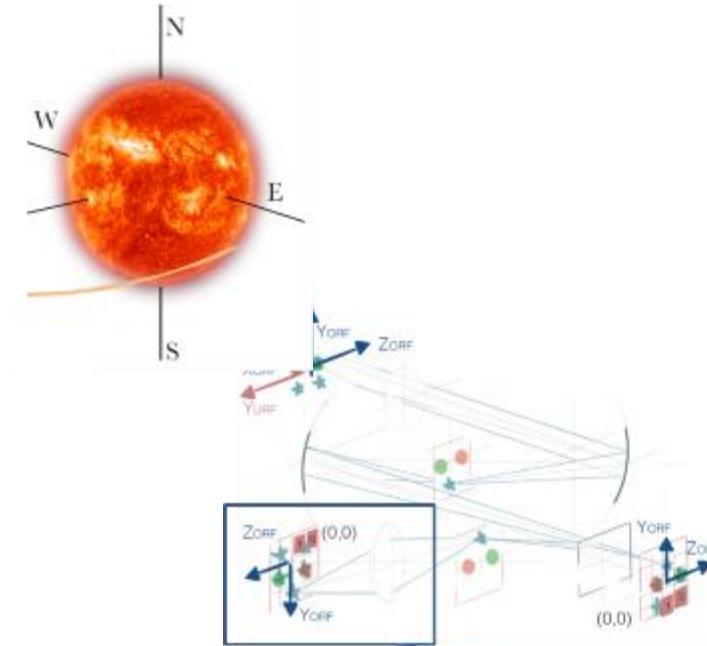
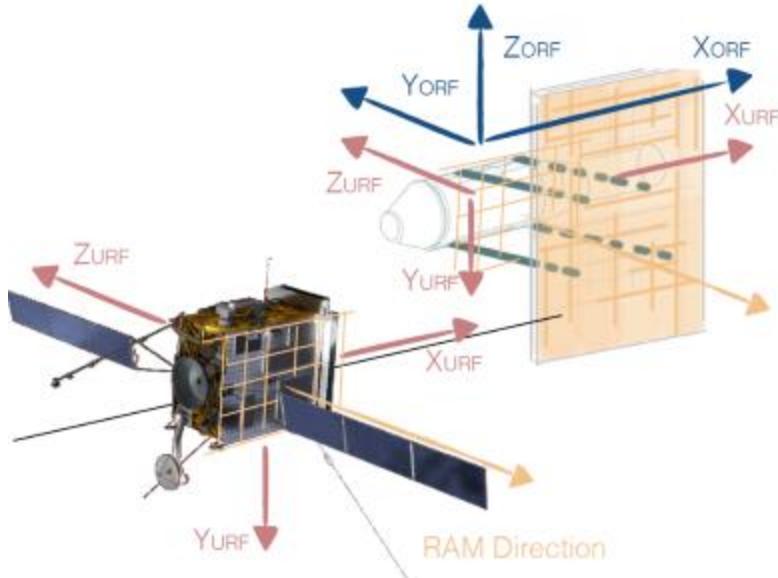
VL Polarimeter Calibration





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Frames of Reference

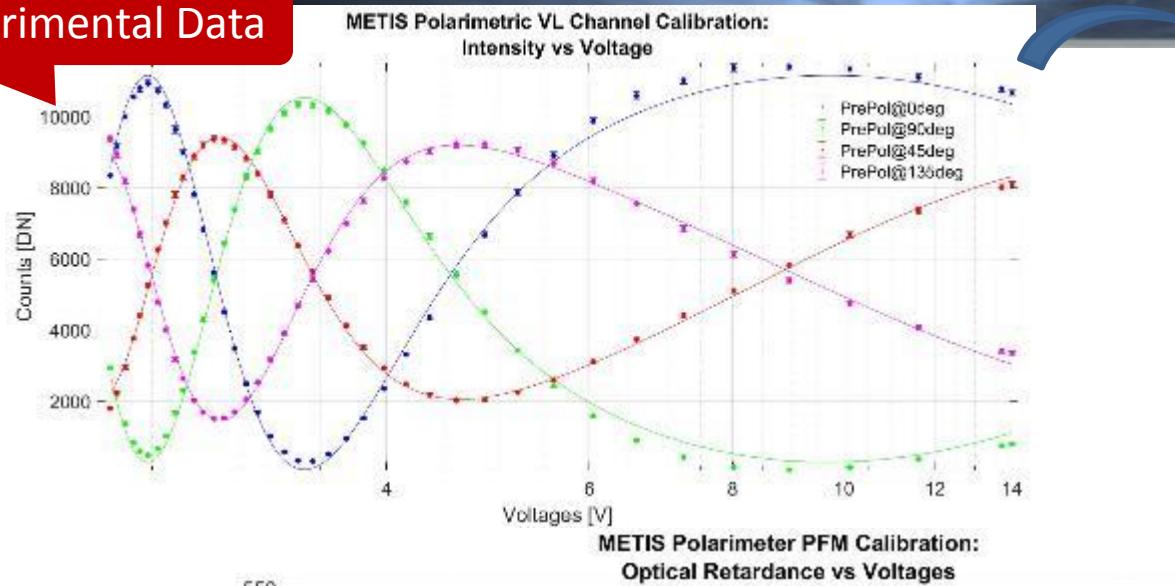




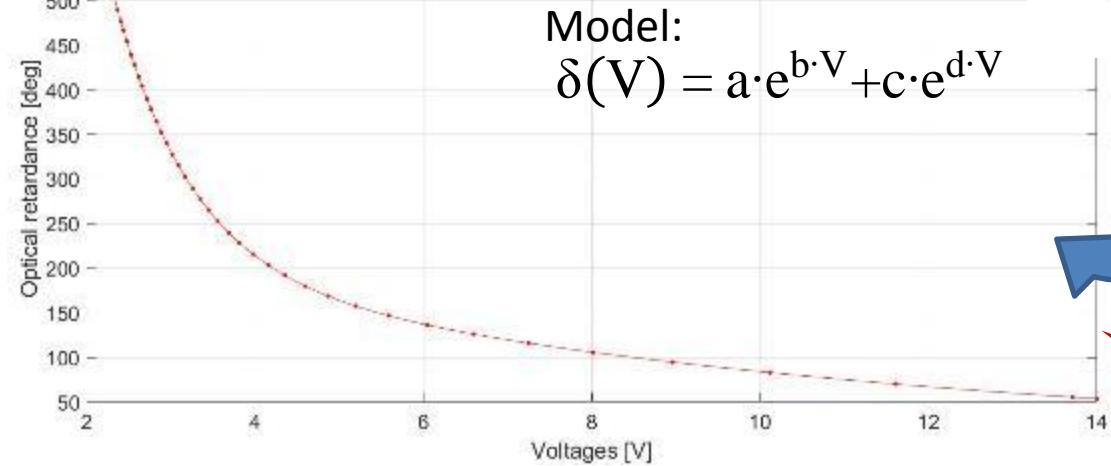
Polarimetry Data Analysis & Results



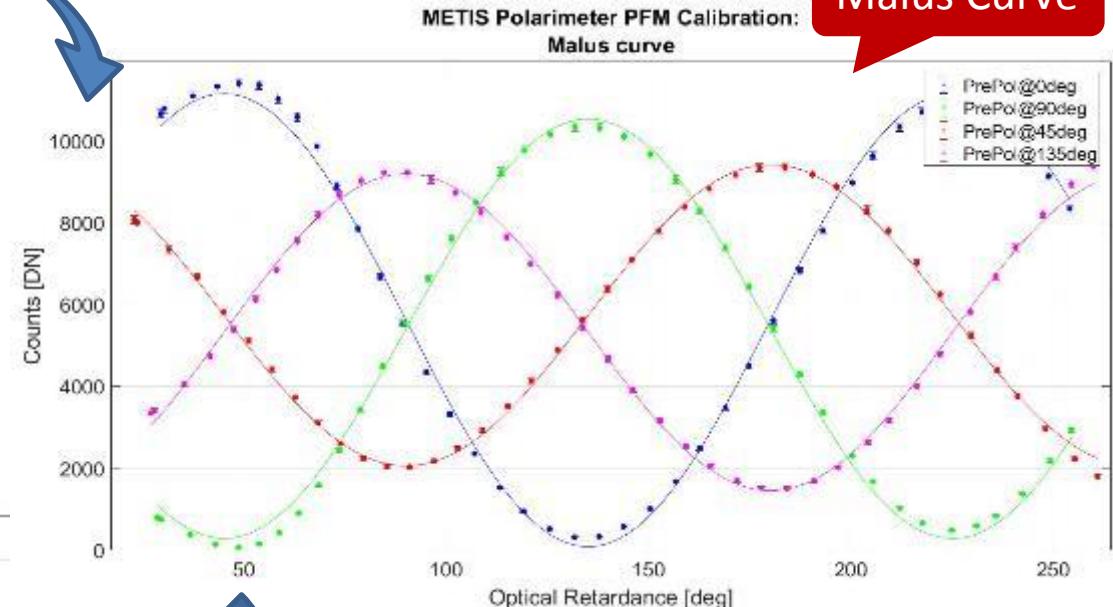
Experimental Data



Model:
$$\delta(V) = a \cdot e^{b \cdot V} + c \cdot e^{d \cdot V}$$



Malus Curve



Retardance-Voltage Curve



Demodulation Tensor Measurement

The demodulation matrix must be measured, and the accuracy to which it is known will determine the accuracy of the measured Stokes parameters.

The incoming light Stokes parameters are obtained by inversion:

$$\begin{matrix} 4 \times 3 \\ m = X \cdot S \end{matrix} \quad \rightarrow \quad S = X^{-1} \cdot m$$

Inversion Problem!!

Method:

Moore-Penrose pseudoinverse → matrix that can act as a partial replacement for the matrix inverse in cases where it does not exist. This matrix is frequently used to solve a system of linear equations when the system does not have a unique solution or has many solutions.



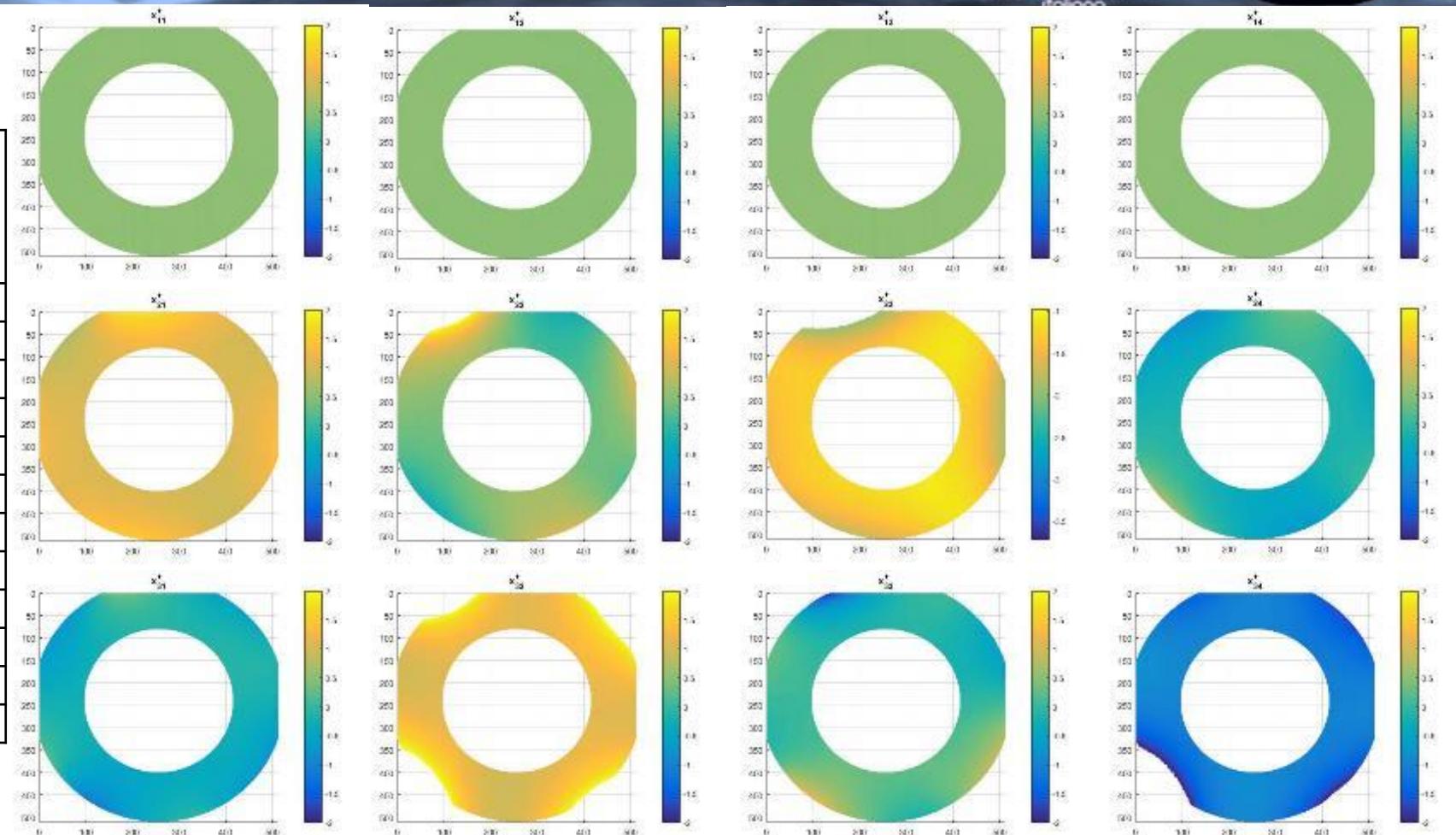
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Demodulation Tensor

Evaluated on the
whole image

Demodulation Matrix Element	Theoretical value	Mean of the computed value	Std of the computed values
x_{11}^+	0.5	0,50	0,00
x_{12}^+	0.5	0,50	0,00
x_{13}^+	0.5	0,50	0,00
x_{14}^+	0.5	0,50	0,00
x_{21}^+	1	1,12	0,15
x_{22}^+	0	0,47	0,38
x_{23}^+	-1	-1,39	0,25
x_{24}^+	0	-0,19	0,24
x_{31}^+	0	-0,29	0,20
x_{32}^+	1	1,35	0,34
x_{33}^+	0	0,06	0,33
x_{34}^+	-1	-1,16	0,31



Cfr. VL channel calibration
Session 11b Thursday 15:00
Marta Casti

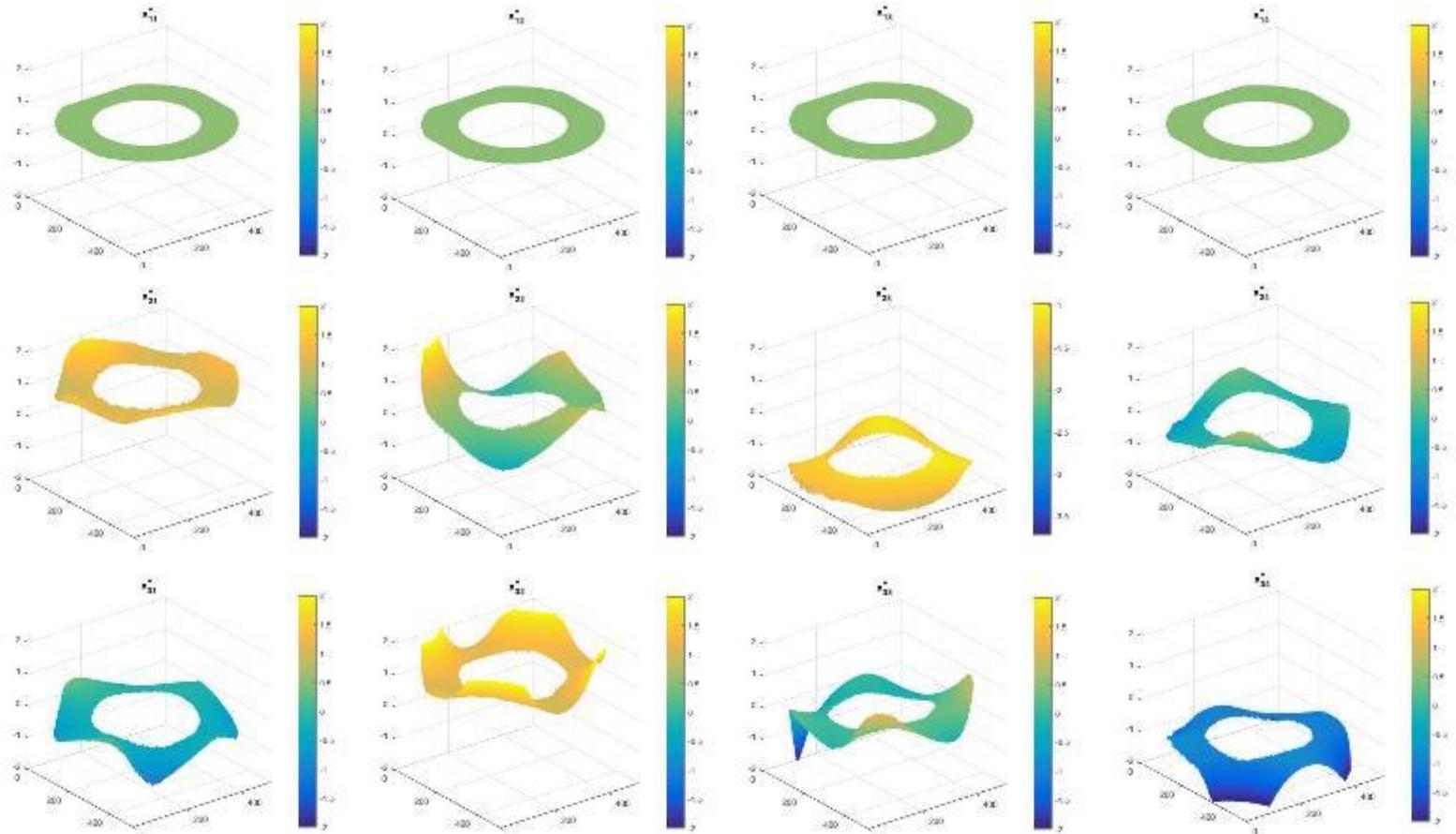


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Demodulation Tensor



$$X^+ = \frac{1}{2} \begin{pmatrix} 1 & 1 & 1 & 1 \\ 2 & 0 & -2 & 0 \\ 0 & 2 & 0 & -2 \end{pmatrix}$$



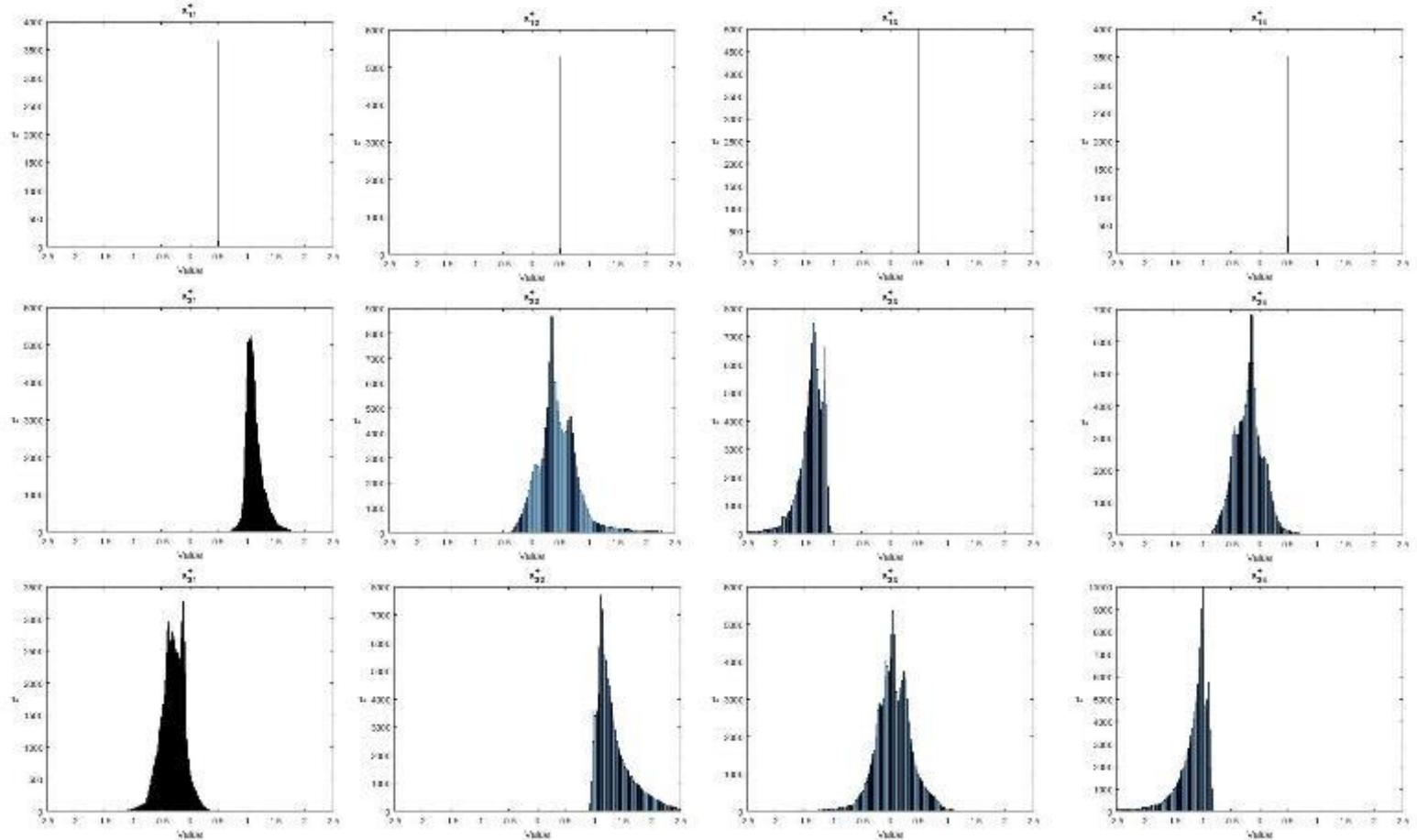


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Demodulation Matrix (parameters variability across the focal plane)



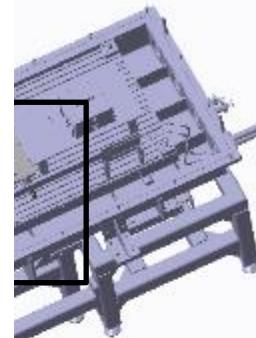
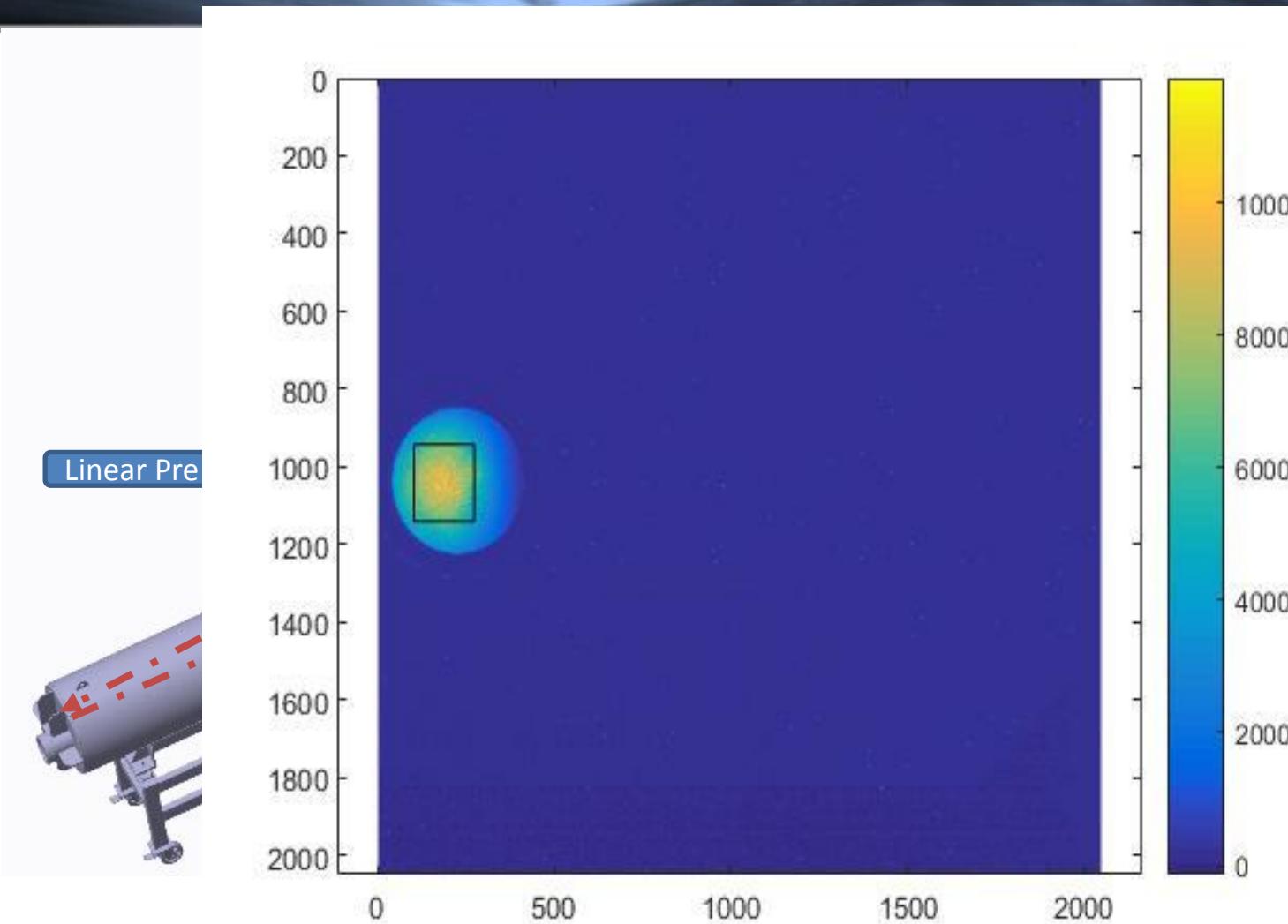
$$X^+ = \frac{1}{2} \begin{pmatrix} 1 & 1 & 1 & 1 \\ 2 & 0 & -2 & 0 \\ 0 & 2 & 0 & -2 \end{pmatrix}$$





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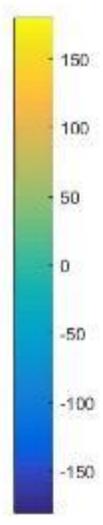
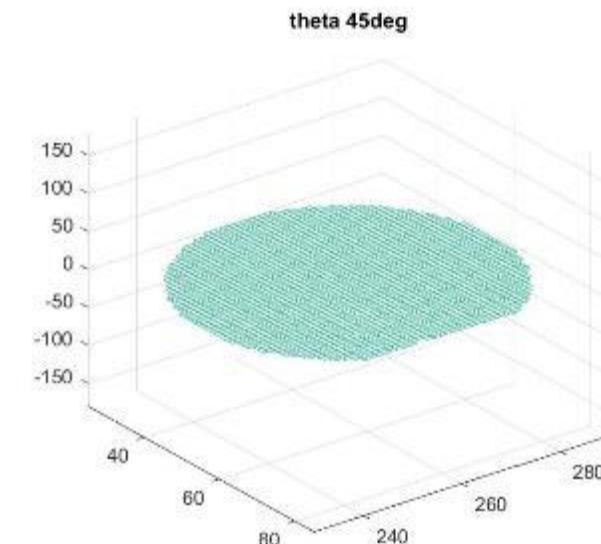
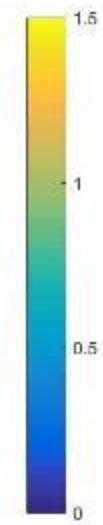
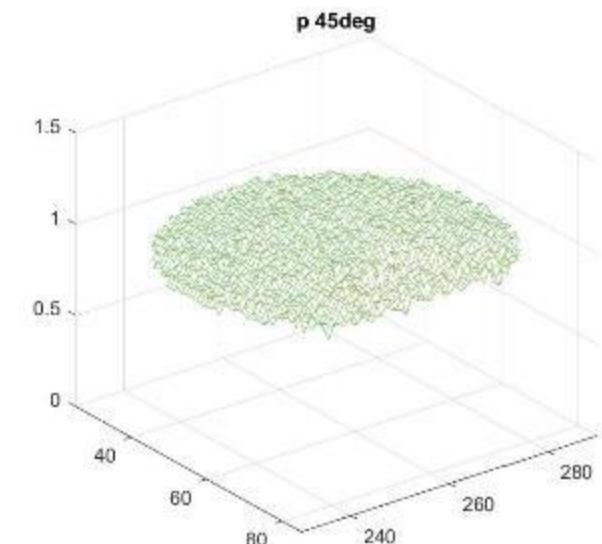
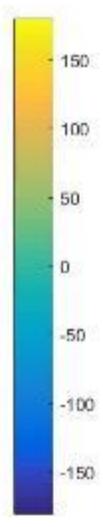
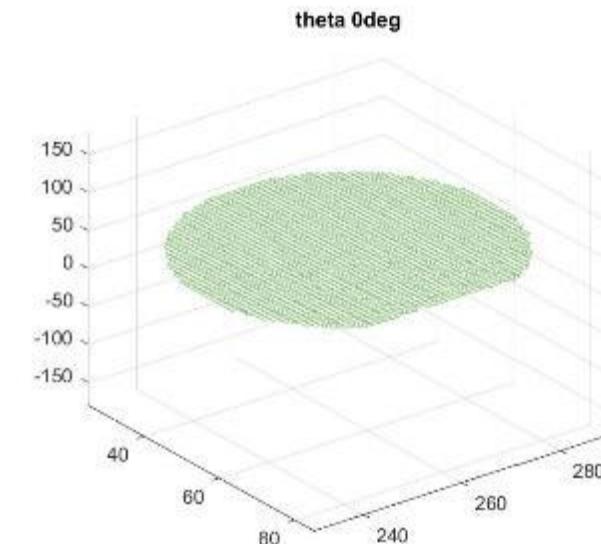
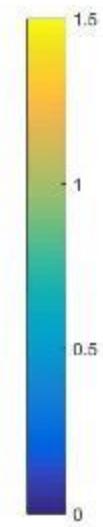
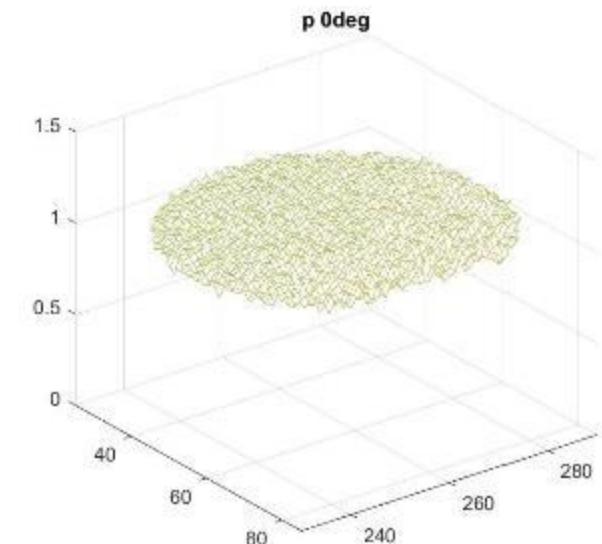
Polarimetry of an Extended Source





SASI Orbiter

Demodulation Matrix Verification



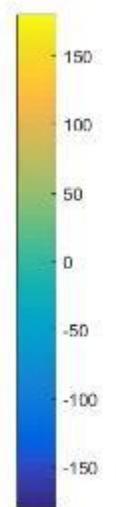
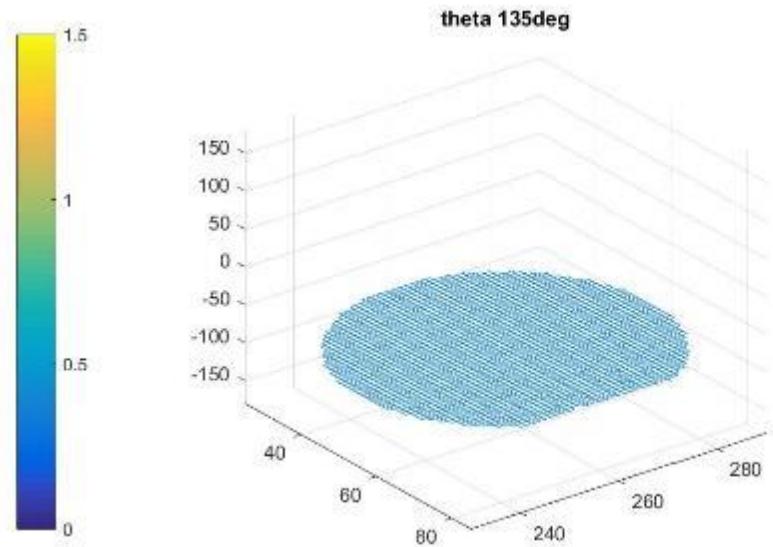
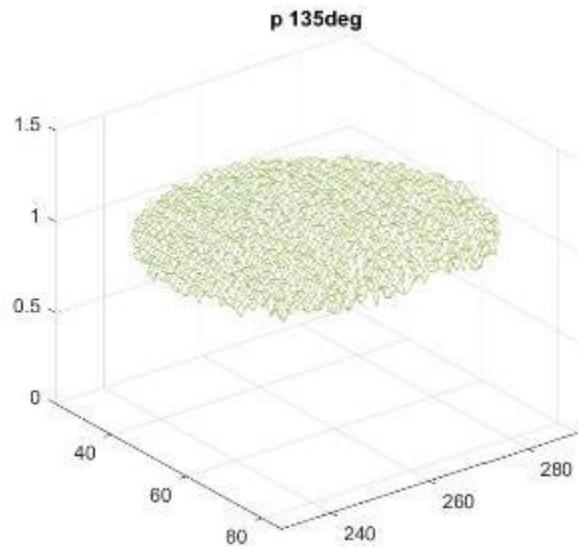
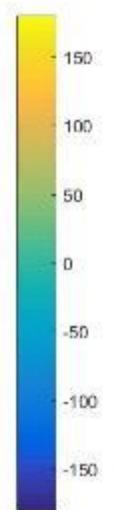
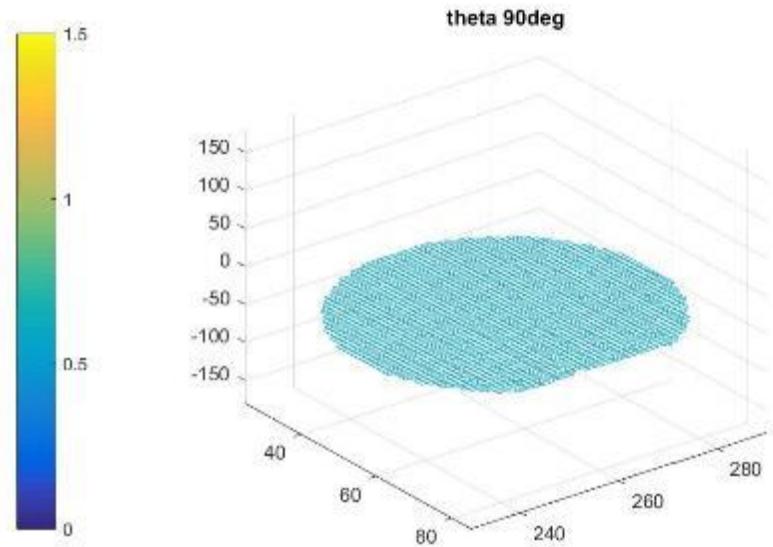
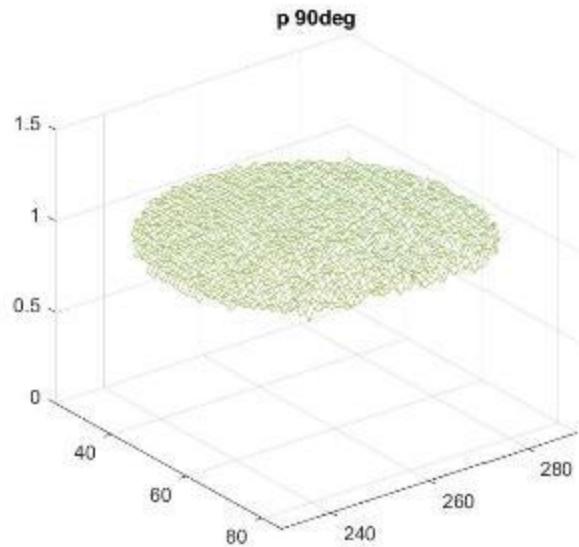


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Demodulation Matrix Verification



metis





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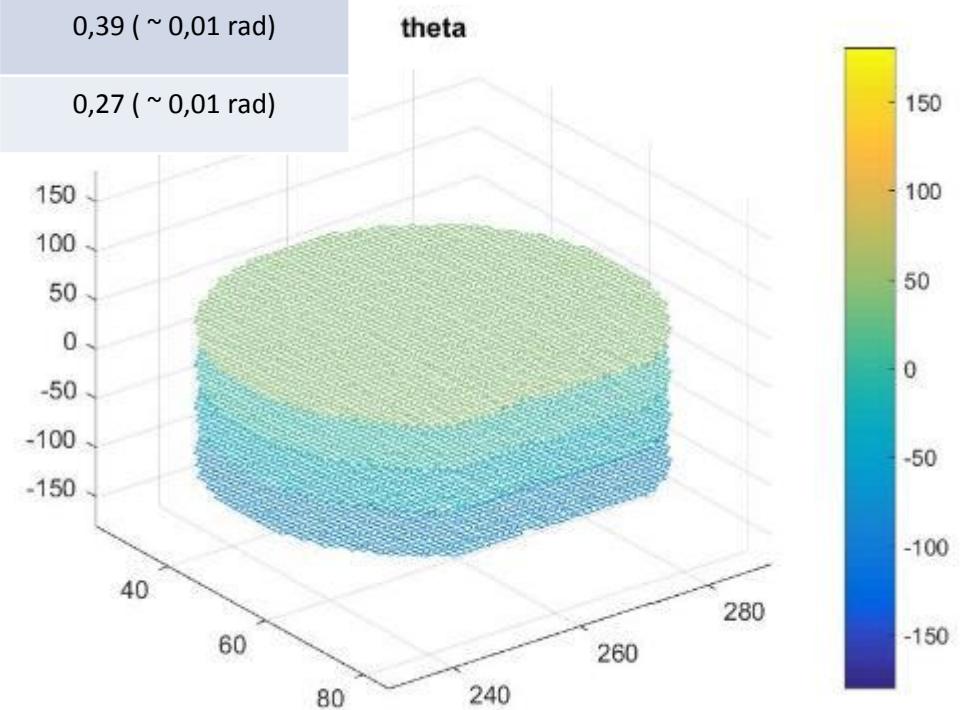


Demodulation Matrix Verification

PrePol Position [deg]	p : Theoretical value	p: Mean of the computed value	p: Standard deviation of the computed value	θ : Theoretical value [deg]	θ : Mean of the computed value [deg]	θ : Standard deviation of the computed value [deg]
0	1	1,04	0,02	45	43,29	0,42 (~ 0,01 rad)
45	1	0,94	0,02	0	-1,04	0,36 (~ 0,01 rad)
90	1	0,99	0,01	-45	-44,01	0,39 (~ 0,01 rad)
135	1	1,00	0,02	90	89,68	0,27 (~ 0,01 rad)

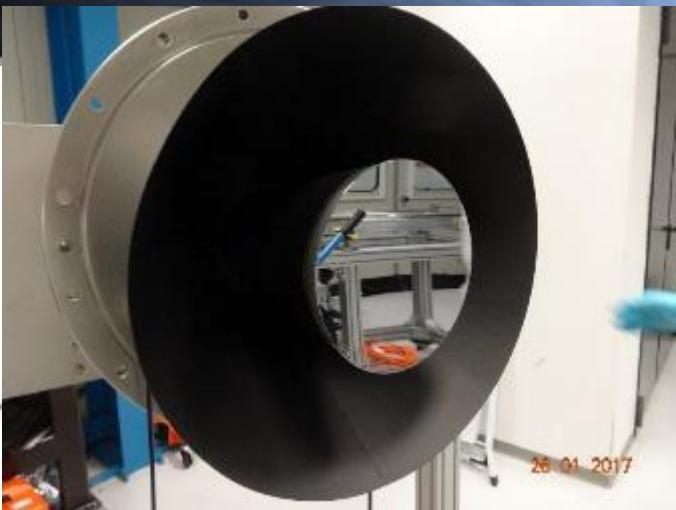
$$\sigma_{th}(p) = \frac{1}{SNR} = \sqrt{\frac{G}{DN}}$$

$$\sigma_{th}(\theta) = \frac{\sigma_{th}(p)}{p}$$

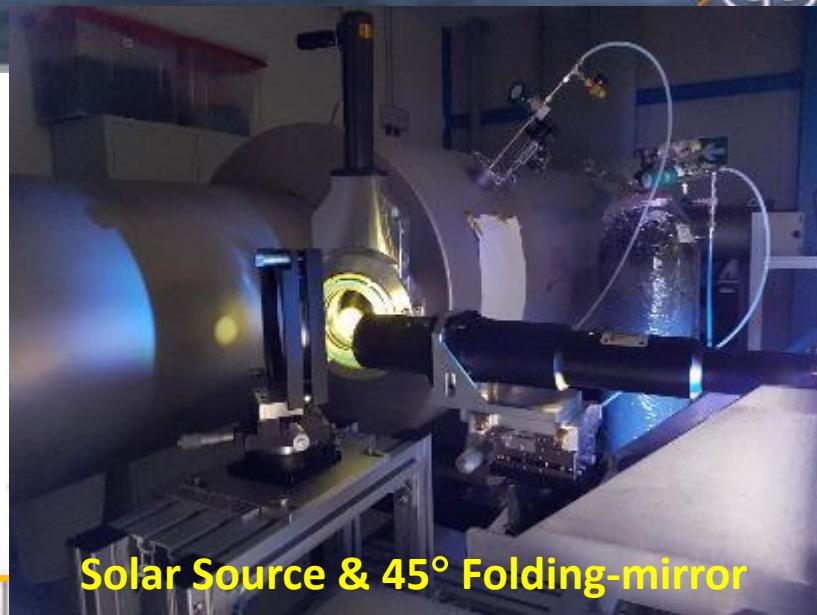




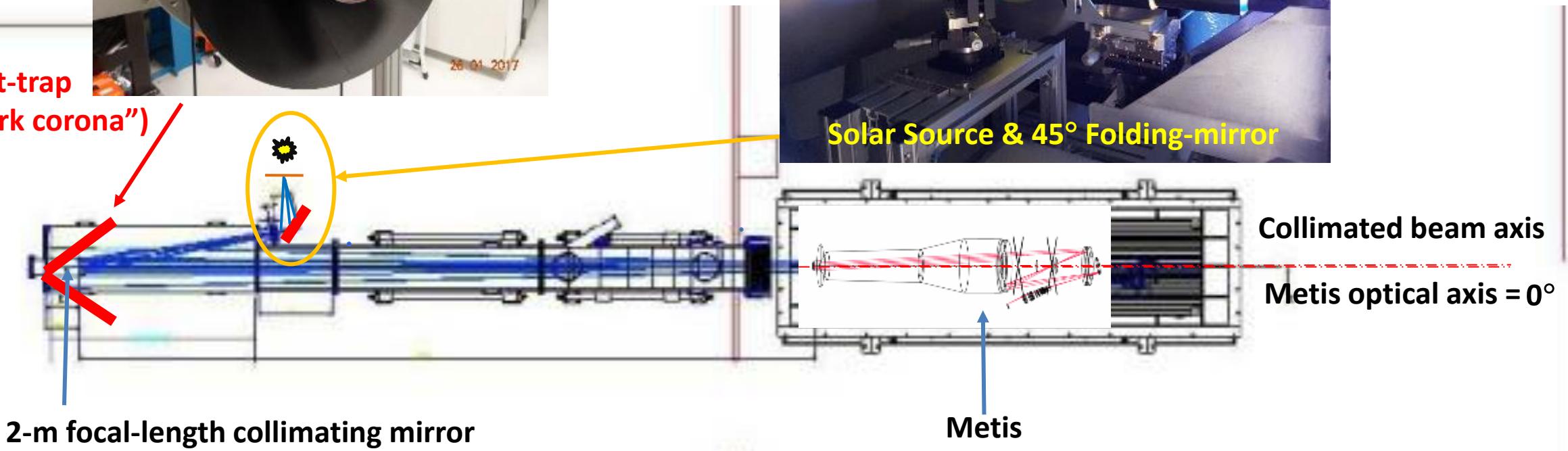
SPOCC Set-up for Stray-Light Measurements



Light-trap
("dark corona")



Solar Source & 45° Folding-mirror



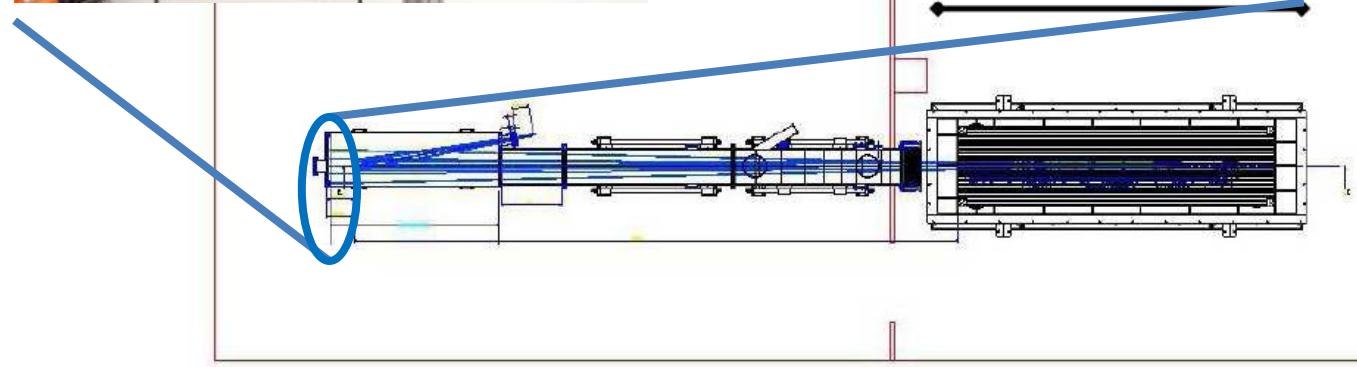
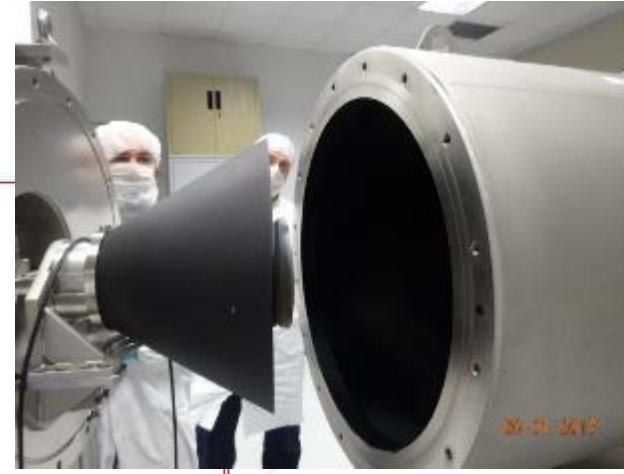
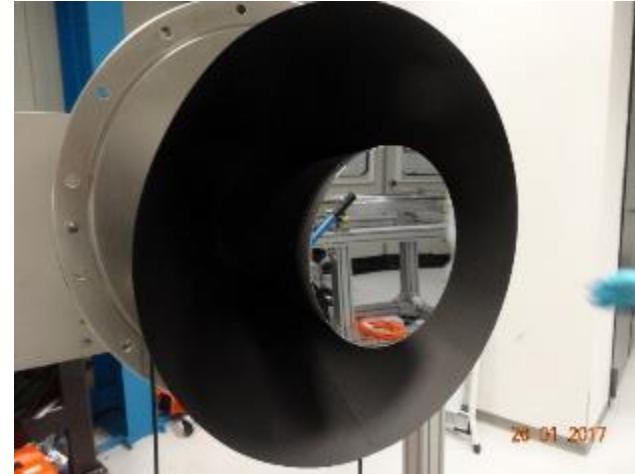
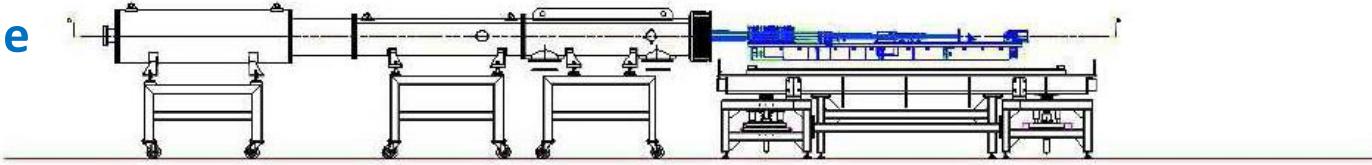


solar orbiter

SPOCC Light-trap



Required level of suppression of the
stray-light: $B_{\text{stray}} / B_{\text{Sun}} \leq 1.\text{e-}9$



27/09/2018

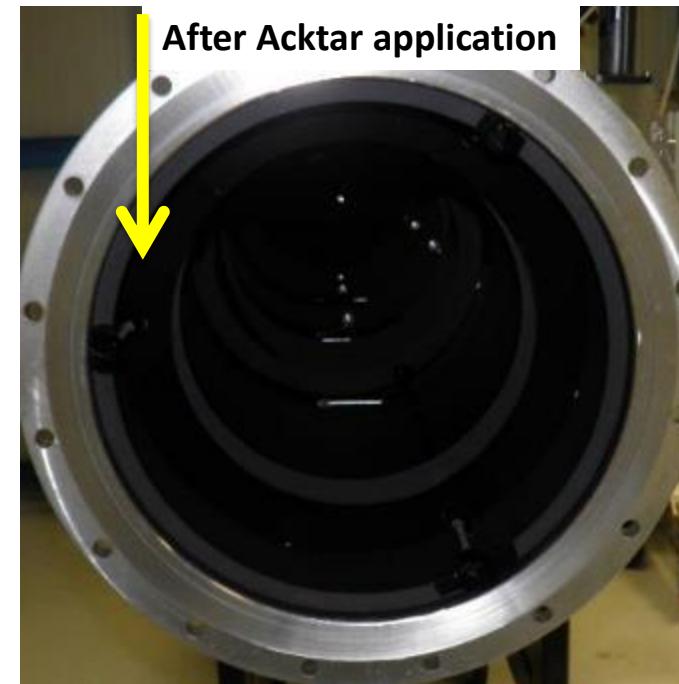
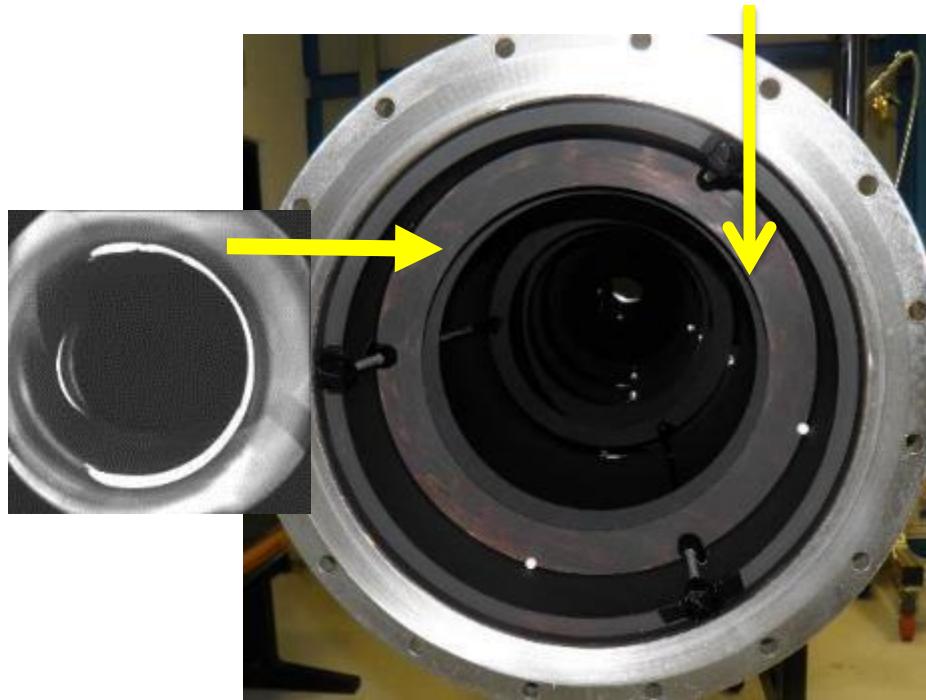


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SOCC Pipeline Baffling



- The bright halo was mainly due to a back reflection of one of the first baffle in the pipeline blackened only by Aeroglaze.

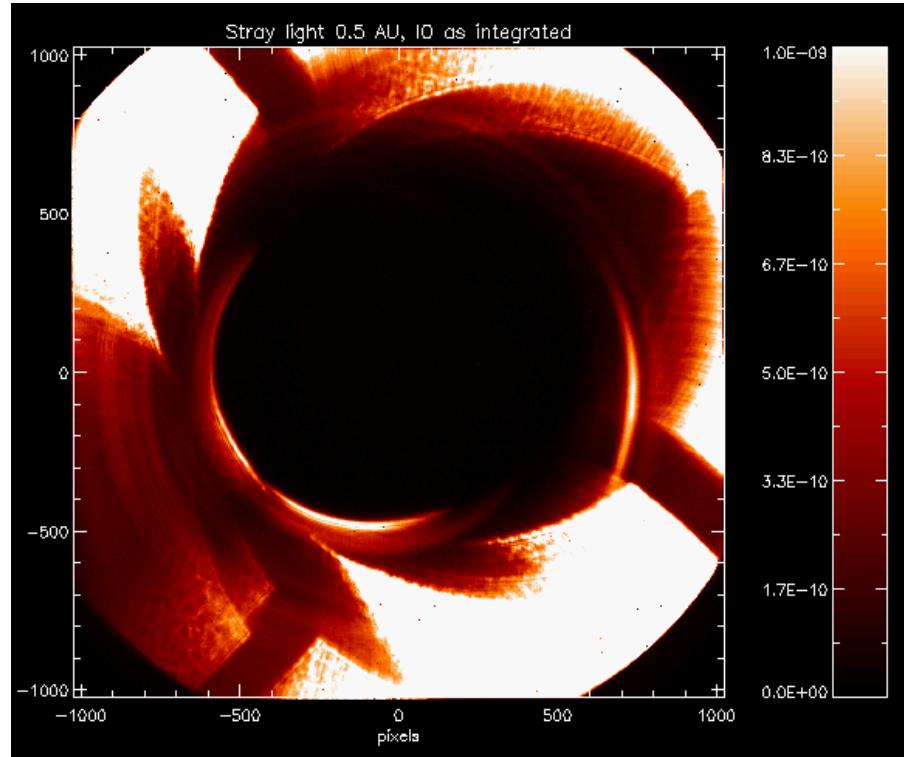




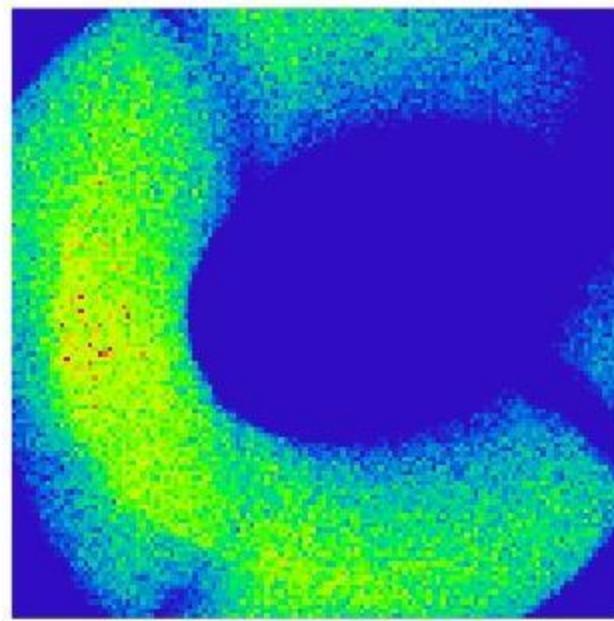
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Stray light at 0.5 AU

before IO position fine tuning



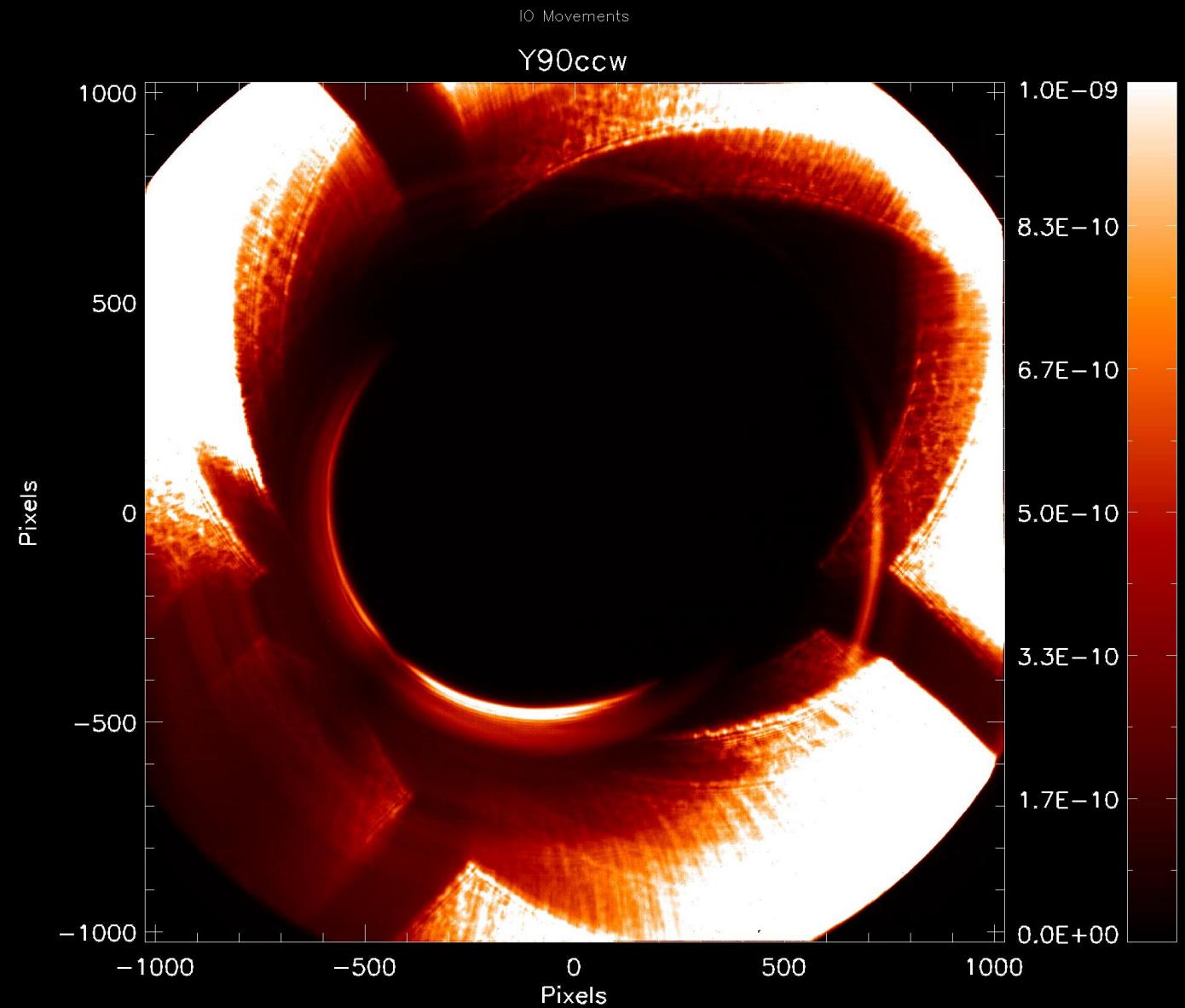
Acquired image



Zemax simulation with off-centered IO



solar o





solar orbiter

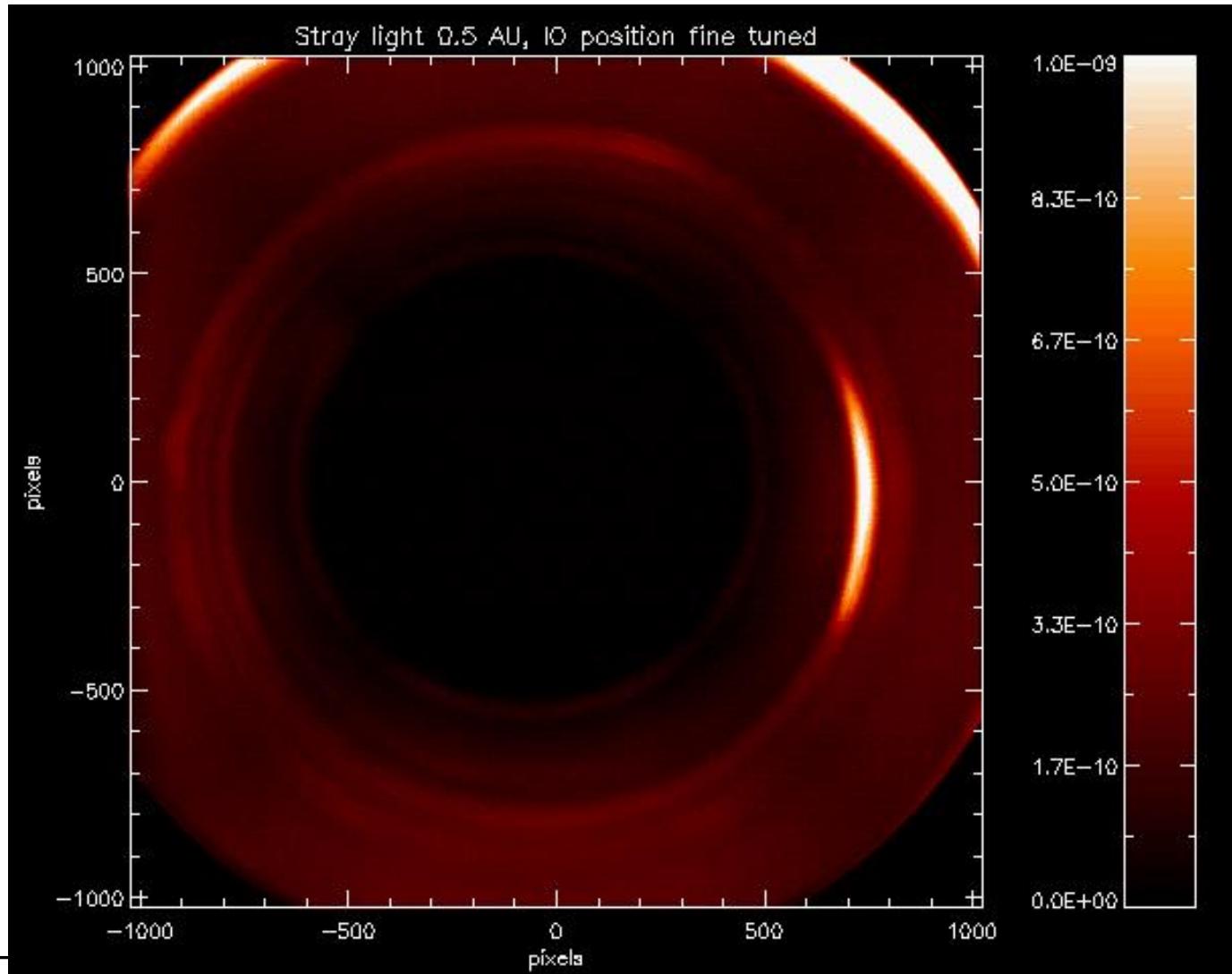


Stray light at 0.5 AU

Level of suppression of the stray-light from the Sun at 0.5 A.U. ("dark corona"):

$$B_{\text{stray}} / B_{\text{Sun}} \leq 5.\text{e-}10$$

(Req.: $B_{\text{stray}} / B_{\text{Sun}} \leq 1.\text{e-}9$)





solar orbiter

Conclusions



- Vignetting function (Flat-Field) ✓ according to ray-trace
- VLDA Photon Transfer Curve ✓ matching component-level calibration
- VL Imaging & Radiometric Performance ✓ within specs
- UV Imaging ✓ PSF $> \times 3$ wrt ray-trace (VL-UV alignment balancing)
- UV Radiometric Performance ✓ less than a factor 2.3 wrt estimated
(due to mirror coating: 54% vs expected 80%)
- VL Polarimeter Performance ✓ within specs
- Stray-light Suppression Performance ✓ within specs



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